

eXtended Finite Element Methods for thin cracked plates with Kirchhoff-Love theory

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September 24, 2009

Abstract

A modelization of cracked plates under bending loads in the XFEM framework is addressed. The Kirchhoff-Love model is considered. It is well suited for very thin plates commonly used for instance in aircraft structures. Reduced HCT and FVS elements are used for the numerical discretization. Two kinds of strategies are proposed for the enrichment around the crack tip with, for both of them, an enrichment area of fixed size (*i.e.* independant of the mesh size parameter). In the first one, each degree of freedom inside this area is enriched with the non-smooth functions that describe the asymptotic displacement near the crack tip. The second strategy consists in introducing these functions in the finite element basis with a single degree of freedom for each one. An integral matching is then used in order to ensure the \mathcal{C}^1 continuity of the solution at the interface between the enriched and the non-enriched areas. Finally, numerical convergence results for these strategies are presented and discussed.

1 Introduction

This paper deals with an adaptation of XFEM (eXtended Finite Element Method) to the computation of thin plates, which present a through-thickness crack. Let us recall that XFEM is a way to introduce the discontinuity across the crack and the asymptotic displacement into the finite element space. It has been initially developed for plane elasticity problems (see [1, 2]) and is now the subject of a wide literature (see for instance [3, 4, 5, 6, 7, 8]).

Nowadays, as far as we know, there are few previous works devoted to the adaptation of XFEM to plate or shell models [9, 10, 11, 12]. In [9, 10], shell models are used: since the near tip asymptotic displacement in this model is unknown, no singular enrichment is formulated in these references. In [11], which deals with cracked shells, the cracked part of the domain is modeled by a three-dimensional XFEM formulation. It is matched with the rest of the domain, formulated with a classical finite element shell model. However, this is not the way we choose to work. In this paper, a plate model is kept on the entire domain, and we consider singular enrichment.

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In [12], the plate model used is the Mindlin-Reissner one. However, in this reference, an important locking effect for thin plates has been detected despite the use of some classical locking-free elements. This suggests that this locking effect is due to the XFEM enrichment. In order to avoid locking, some special treatments are mandatory, for example selective reduced integration in the case of the QUAD 4 element (see [13]). But these treatments are not straightforwardly adaptable to an enrichment by nonsmooth functions, which are not polynomials.

The most popular models for the small deformations of plates are the so called Mindlin-Reissner and Kirchhoff-Love ones. The Mindlin-Reissner model is convenient for moderately thin plates. It takes into account the transverse shear strain but finite element approximations are subject to the so-called shear locking phenomenon for very thin plates. The Kirchhoff-Love model, which is not subject to the shear locking phenomenon, provides a realistic description of the displacement of thin plates, and especially very thin ones, since it is the limit model when the thickness vanishes (see [14]). It has already been used for the purpose of fracture mechanics (for instance, see [15]). Moreover, for through-thickness cracks, the limit of the energy release rate of the three-dimensional model can be expressed with the Kirchhoff-Love model solution (see [16] and [17]).

The Kirchhoff-Love model corresponds to a fourth order partial differential equation. Consequently, a conformal finite element method needs the use of \mathcal{C}^1 (continuously differentiable) finite elements. To avoid too costly elements we consider the reduced Hsieh-Clough-Tocher triangle (reduced HCT) and Fraeijs de Veubeke-Sanders quadrilateral (reduced FVS) (see [18]). They lead to a satisfactory theoretical accuracy, with a reasonable computational cost. Let us also remark that, for this plate model, the crack tip asymptotic bending displacement is well-known for an isotropic plate: it corresponds to the bilaplacian problem one.

Now, concerning the specific XFEM enrichment, a “jump function” (or Heaviside function) is used in order to represent the discontinuity due to the crack. Following the ideas already presented in [4], we propose two strategies for the crack tip enrichment. In both of them, an enrichment area of fixed size is defined, centered on the crack tip. In the first strategy, each node contained in the enrichment area has all its degrees of freedom enriched with the crack tip asymptotic bending. For the second strategy, the asymptotic bending displacements are introduced in a global way. Then a matching condition is needed in order to ensure the continuity of the displacement and its derivatives across the interface between the enrichment area and the remaining part of the domain.

Along this paper, it will be underlined that our XFEM formulation presents a sense of optimality, both about accuracy and computational cost, since they are nearly equal to those of a classical finite element method on a regular non-cracked problem.

This paper is organized as follows. Section 2 describes the model problem. Section 3 is devoted to some aspects of the finite element discretization of the Kirchhoff-Love model. In Section 4, the enrichment strategies are presented: they are evaluated on two test problems in Section 5.

2 The Model Problem

2.1 Notations and variational formulation

Let us consider a thin plate, *i.e.* a plane structure for which one dimension, called the thickness, is very small compared to the others. For this kind of structures, starting from *a priori* hypotheses on

the expression of the displacement fields, a two-dimensional problem is usually derived from the three-dimensional elasticity formulation by means of integration along the thickness. Then, the unknown variables are set down on the mid-plane of the plate. In all the following, this mid-plane will be denoted by Ω . It is an open subset of \mathbb{R}^2 . So, in a three-dimensional cartesian referential, the plate is the set

$$\{ (x_1, x_2, x_3) \in \mathbb{R}^3, (x_1, x_2) \in \Omega \text{ and } x_3 \in]-\varepsilon; \varepsilon[\} .$$

The x_3 coordinate corresponds to the transverse direction. All the mid-plane points have their third coordinate equal to 0 and the thickness is 2ε (see Fig. 1). Finally, we assume that the plate has a through-thickness crack (see Fig. 1) and that the material is isotropic of Young's modulus E and Poisson's ratio ν .

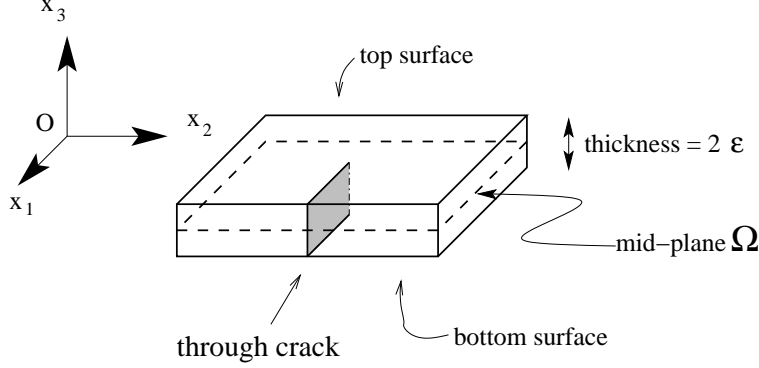


Figure 1: Cracked thin plate (the thickness is oversized for the sake of clarity).

In plate theory, it is usual to consider the following approximation of the three-dimensional displacements for $(x_1, x_2, x_3) \in \Omega \times]-\varepsilon, \varepsilon[$

$$\begin{cases} u_1(x_1, x_2, x_3) = \bar{u}_1(x_1, x_2) + x_3 \psi_1(x_1, x_2), \\ u_2(x_1, x_2, x_3) = \bar{u}_2(x_1, x_2) + x_3 \psi_2(x_1, x_2), \\ u_3(x_1, x_2, x_3) = u_3(x_1, x_2). \end{cases} \quad (1)$$

In these expressions, \bar{u}_1 and \bar{u}_2 are the membrane displacements of the mid-plane points while u_3 is the deflection, ψ_1 and ψ_2 are the section rotations. In the case of an homogeneous isotropic material, the variational plate model splits into two independent problems: the first, called the membrane problem, deals only with the membrane displacements, while the second, called the bending problem, concerns the deflection and the rotations. The membrane problem corresponds to the classical plane elasticity problem and has been already treated in many references (see for instance [4, 5]). In this paper, we shall only adress the bending problem.

For reasons mentioned in the introduction, we consider the Kirchhoff-Love model, which can be seen as a particular case of (1), as it is obtained by introducing the Kirchhoff-Love assumptions, which read as

$$\psi = -\nabla u_3 \Leftrightarrow \begin{cases} \psi_1 = -\partial_1 u_3, \\ \psi_2 = -\partial_2 u_3, \end{cases} \quad (2)$$

where, the notation ∂_α stands for the partial derivative with respect to x_α (for $\alpha = 1, 2$). A first consequence of this relation is that the transverse shear strain is identically zero, which avoids the shear locking problem. A second consequence of (2) is that the section rotation only depends

on the transverse displacement. It means that this displacement is the only unknown function for the bending problem. For convenience, it will be denoted by u all along the following of this paper. So, in the Kirchhoff-Love framework and for a pure bending problem, the three-dimensional displacement reads as

$$\begin{cases} u_1(x_1, x_2, x_3) &= -x_3 \partial_1 u(x_1, x_2), \\ u_2(x_1, x_2, x_3) &= -x_3 \partial_2 u(x_1, x_2), \\ u_3(x_1, x_2, x_3) &= u(x_1, x_2). \end{cases}$$

For the sake of simplicity, we assume the plate to be clamped on its boundary, except along the crack faces where a traction free condition is considered. The plate is subject to a volume force f of coordinates (f_1, f_2, f_3) , and two surface forces, say g^+ and g^- , applied on the top and bottom surfaces. The variational formulation (or virtual work formulation) of the Kirchhoff-Love model reads as

$$\begin{cases} \text{Find } u \in H_0^2(\Omega) \text{ such that for any } v \in H_0^2(\Omega) \\ \int_{\Omega} \frac{2E\varepsilon^3}{3(1-\nu^2)} [(1-\nu) \partial_{\alpha\beta}^2 u + \nu \Delta u \delta_{\alpha\beta}] \partial_{\alpha\beta}^2 v \, dx = \int_{\Omega} [F v + M_{\alpha} \partial_{\alpha} v] \, dx \end{cases} \quad (3)$$

where $F = g_3^+ + g_3^- + \int_{-\varepsilon}^{\varepsilon} f_3 \, dx_3$ is the resulting transverse loading and $M_{\alpha} = \varepsilon(g_{\alpha}^+ - g_{\alpha}^-) + \int_{-\varepsilon}^{\varepsilon} x_3 f_{\alpha} \, dx_3$ is the resulting moment loading. Moreover $\delta_{\alpha\beta}$ stands for the Kronecker's symbol and the summation convention over repeated indices is adopted, Greek indices varying in $\{1, 2\}$. Finally, $H_0^2(\Omega)$ is the classical Sobolev space, the definition of which can be found in [19] for instance.

2.2 Asymptotic displacement near the crack tip and Fracture Modes

In the Kirchhoff-Love plate model, there are two fracture modes. Applying a symmetric bending leads to the first fracture mode, while applying an anti-symmetric bending or a transverse shear, leads to the second one (see Fig. 2).

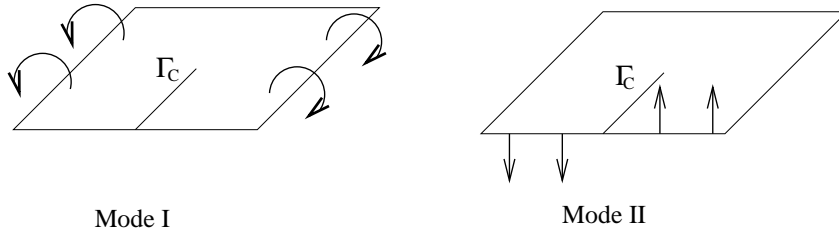


Figure 2: Fracture modes for Kirchhoff-Love bending model (Γ_C is the cracked part of the boundary). Left: a symmetric bending leads to mode I. Right: a shear bending leads to mode II.

To characterize them, let us recall that the governing equation related to the bending variational problem (3) reads

$$\frac{2E\varepsilon^3}{3(1-\nu^2)} \Delta^2 u = F + \partial_{\alpha} M_{\alpha} \quad (4)$$

on the mid-plane Ω . It is a bilaplacian problem for which the asymptotic behavior near the crack-tip is well-known (see [20]). So, close to the crack tip, the displacement may be written as $u = u_r + u_s$, where u_r stands for the regular part of the transverse displacement and belongs to $H^3(\Omega)$. The nonsmooth part u_s belongs to $H^{5/2-\eta}(\Omega)$, for any $\eta > 0$, and reads

$$u_s(r, \theta) = A r^{3/2} \left[K_1 \left(\frac{\nu + 7}{3(\nu - 1)} \cos \frac{3}{2}\theta + \cos \frac{\theta}{2} \right) + K_2 \left(\frac{3\nu + 5}{3(\nu - 1)} \sin \frac{3}{2}\theta + \sin \frac{\theta}{2} \right) \right] \quad (5)$$

in polar coordinates relatively to the crack tip and with $A = \frac{\sqrt{2}}{2} \frac{1 - \nu^2}{E\varepsilon(3 + \nu)}$. The scalar coefficients K_1 and K_2 are the so-called "Stress Intensity Factors". They are widely used in fracture mechanics for crack propagation.

Remark 1 *The singular modes of the Kirchhoff-Love theory, which are presented in Fig. 2, are different from the ones of the well-known three-dimensional elasticity theory. Some interesting details on this subject are given in [21].*

3 Finite element approximation of the Kirchhoff-Love model

3.1 Choice of the finite element discretization

Let us introduce now the finite element discretization of the variational formulation (3). As usual, a finite dimensional approximation space V^h of the continuous space of solutions $H_0^2(\Omega)$ is built. In order to have a conformal method, *i.e.* $V^h \subset H_0^2(\Omega)$, the functions of V^h must be continuously differentiable, which needs the use of \mathcal{C}^1 finite elements. Among the available elements having this regularity (see [18]), the reduced HCT triangles and FVS quadrangles are of particular interest. For the HCT (resp. FVS) element, the triangle (resp. quadrangle) is divided into three (resp. four) sub-triangles (see Fig. 3). The basis functions of these elements are P_3 polynomials on each sub-triangle and matched \mathcal{C}^1 across each internal edge. In addition, to decrease the number of degrees of freedom (dof), the normal derivative is assumed to vary linearly along the external edges of the elements (this assumption does not hold on the internal edges). Finally, both for triangles and quadrangles, there are only three dof on each node: the value of the function and its first derivatives. So, these elements have the two following properties:

1. With the reduction, the computational cost is limited to three dof for each node of the mesh, like a classical Mindlin-Reissner element for which each node has also three dof (the deflection and the two section rotations).
2. For regular problems (see [22]), the theoretical error is in $O(h)$ for the *energy norm* and $O(h^2)$ for the *displacement norm*, where h stands for the mesh parameter.

Then, the reduced HCT or FVS elements and the standard Mindlin elements have the same features as far as numerical cost and accuracy are concerned.

Remark 2 *One can also notice that reduced HCT/FVS elements have the same degrees of freedom. Consequently, it is possible to use meshes made of triangles and quadrangles at the same time.*

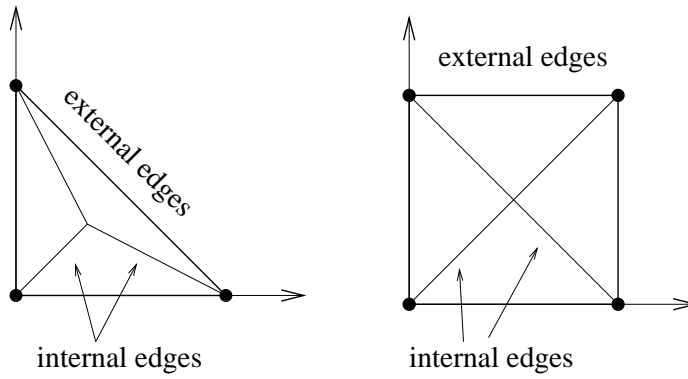


Figure 3: HCT triangle and FVS quadrangle. Location of degrees of freedom and sub-triangles.

The shape functions of the reduced HCT and FVS are presented in Fig. 4 and 5. The shape functions corresponding to the Lagrange dof, and to the x_1 -derivative dof are shown (the remaining one corresponding to the x_2 -derivative dof can be obtained by symmetry). Note that the expression of the shape functions is rather complex because it is a polynomial of degree three in each sub-triangle. The implementation used in Getfem++ [23] uses the expression of the shape functions of the (not reduced) HCT and FVS elements on the reference element (complete expressions are freely available in the Getfem++ sources). The reduction operation is performed during the transformation from the reference element to the real one.

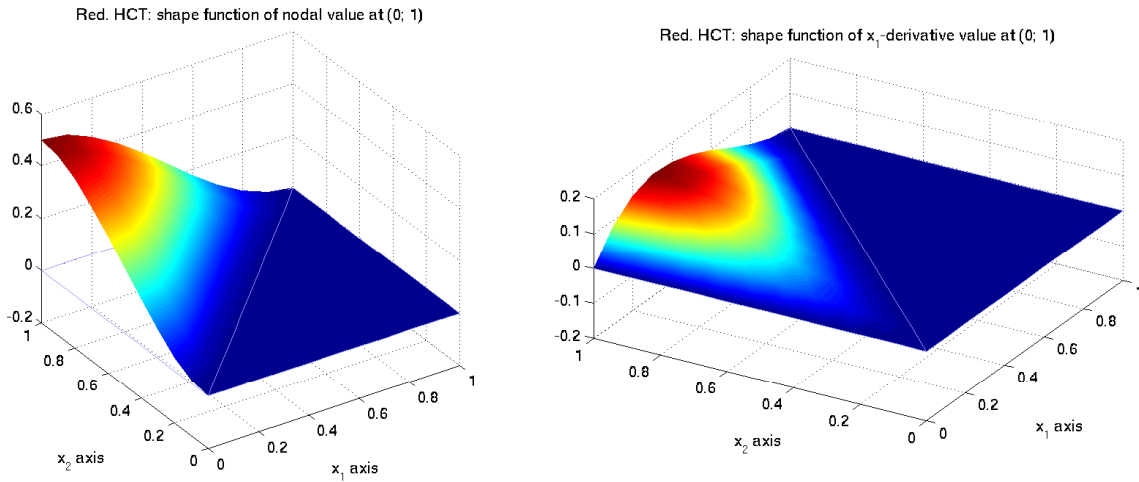


Figure 4: Shape functions of the reduced HCT. Left: nodal value. Right: x_1 -derivative value.

3.2 Integral matching

As mentioned in the introduction, one of the strategies, which will be presented further, needs to match the enriched subdomain with the non-enriched one. So we shall have to carry out calculations in the case of a domain splitted into two subdomains, including an *integral matching* at the subdomains boundary.

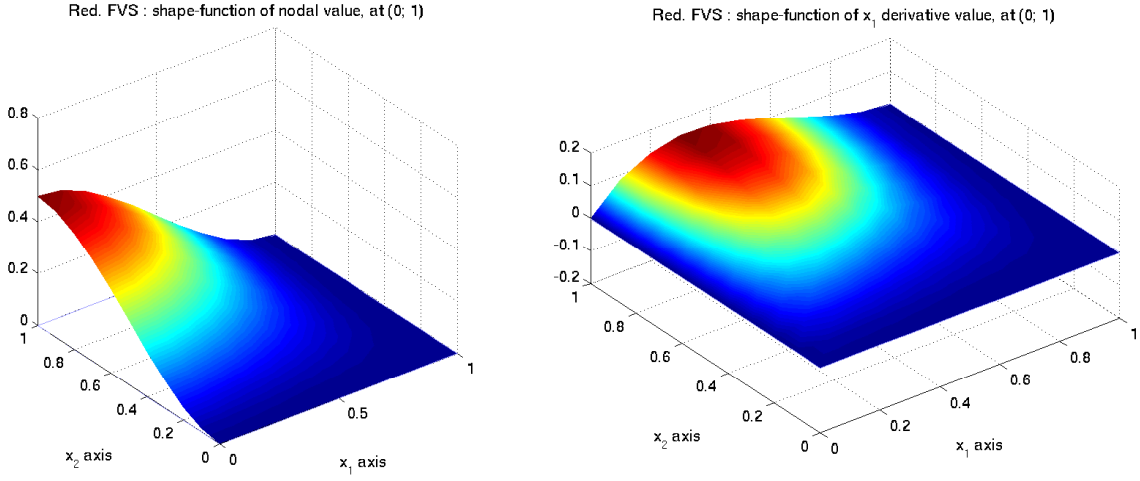


Figure 5: Shape functions of the reduced FVS. Left: nodal value. Right: x_1 -derivative value.

Some theoretical results exist on the convergence of the mortar methods for fourth order problems. For instance, in [24, 25], it is shown that the use of mortar methods with reduced HCT/FVS elements can achieve an error in $O(h)/O(h^2)$ for the energy/displacement norms. Nevertheless, as far as we know, no numerical experiment checking the accuracy of such a method has been published. Moreover, we don't use exactly the method described in [24, 25]. So, before introducing our enrichment strategy on a cracked domain, it appeared meaningful to test our integral matching conditions first.

Let us consider the problem (4) without crack. The loading F is such that the exact solution is $u(x_1, x_2) = \sin(10(x_1 + x_2))$. Let us observe that it also implies non-homogeneous Dirichlet boundary conditions. In a first step, we compute the solution to this problem on structured and non-structured meshes with the reduced HCT/FVS elements. This enables to check that the errors in energy and displacement norms are in accordance with what is expected. It also establishes a reference error level which will be compared to the mortar method one.

Then, we consider the domain is splitted into two subdomains, say Ω_1 and Ω_2 , with matching triangulations. It means that, on both sides of the boundary Γ between the two subdomains, the nodes are located at the same place, see Fig. 6. It is identical to the so-called "*mortar method with matching triangulations*", in [24]. The matching conditions, which involve the values of the unknowns, say u_i on each part Ω_i , are introduced in order to ensure the continuity of the function and its normal derivative. The following relations were chosen at this aim

$$\begin{aligned}
 \int_{\Gamma} u_1 \lambda d\Gamma &= \int_{\Gamma} u_2 \lambda d\Gamma & , \quad \forall \lambda \in \Lambda, \\
 \int_{\Gamma} \partial_n u_1 \mu d\Gamma &= \int_{\Gamma} -\partial_n u_2 \mu d\Gamma & , \quad \forall \mu \in M,
 \end{aligned}
 \tag{6}$$

where Λ and M are appropriate multiplier spaces. Let us notice that the change of sign in front of the normal derivative $\partial_n u_i$ (second equation of (6)) is due to the fact that the outside normal vector has an opposite sign whether it refers to Ω_1 or Ω_2 . In our numerical experiments, the multipliers spaces Λ and M are spaces of piecewise polynomials of degree 2 and 1, respectively.

The numerical results for the mortar method, on structured, non-structured, triangular and

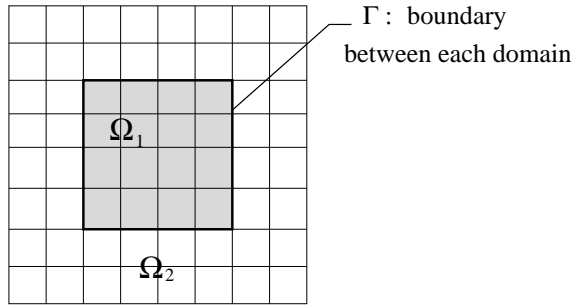


Figure 6: Matching decomposition of the domain Ω .

quadrangular meshes, are given in Fig. 7, where they are compared with the classical method. It shows that our integral matching does not affect the accuracy: the convergence curves are nearly identical in all cases.

4 XFEM enrichments

4.1 Basics

As in previous papers [1, 4, 5], the displacement discontinuity across the crack is represented using a Heaviside-like function, which is multiplied by the finite element shape functions. An enrichment area of fixed size is defined and some nonsmooth functions are added inside this area. Then, the global unknown function may be written as:

$$u^h = \sum_i a_i \varphi_i + \sum_{j \in J} b_j H \varphi_j + \sum_{k \in K} c_k f_k, \quad (7)$$

where function H is equal to $+1$ on one side of the crack and -1 on the other and φ_i stands for the reduced HCT/FVS shape functions. The set J denotes the degrees of freedom whose associated shape functions support is completely crossed by the crack (see Fig. 8). The set K and functions f_k are connected with the crack tip enrichment; their exact meaning and expressions will be detailed further, after next section dealing with a numerical problem we met and the way we chose to overcome it.

4.2 An incompatibility between H-enrichment and FVS element

It appears a kind of numerical incompatibility between the Heaviside-like enrichment and the FVS element, reduced or not. If a straight crack crosses two adjacent edges of a quadrangle, then a *non-invertible* linear system may result. The reason is there exists a function, in the FVS functions space, that is completely zero on two adjacent sub-triangles of the quadrangle and not on the two others (see Fig. 9). Consequently, if the quadrangle is cut along those two sub-triangles (see Fig. 10, left), the regular degrees of freedom associated with the opposite node are linear combinations of the H -enriched ones. This is not a proof that the resulting matrix will be singular, but at least an insight. In our numerical applications, with not too coarse meshes, the linear system was always singular. An example of such a mesh is given in Fig. 11. In addition, this problem

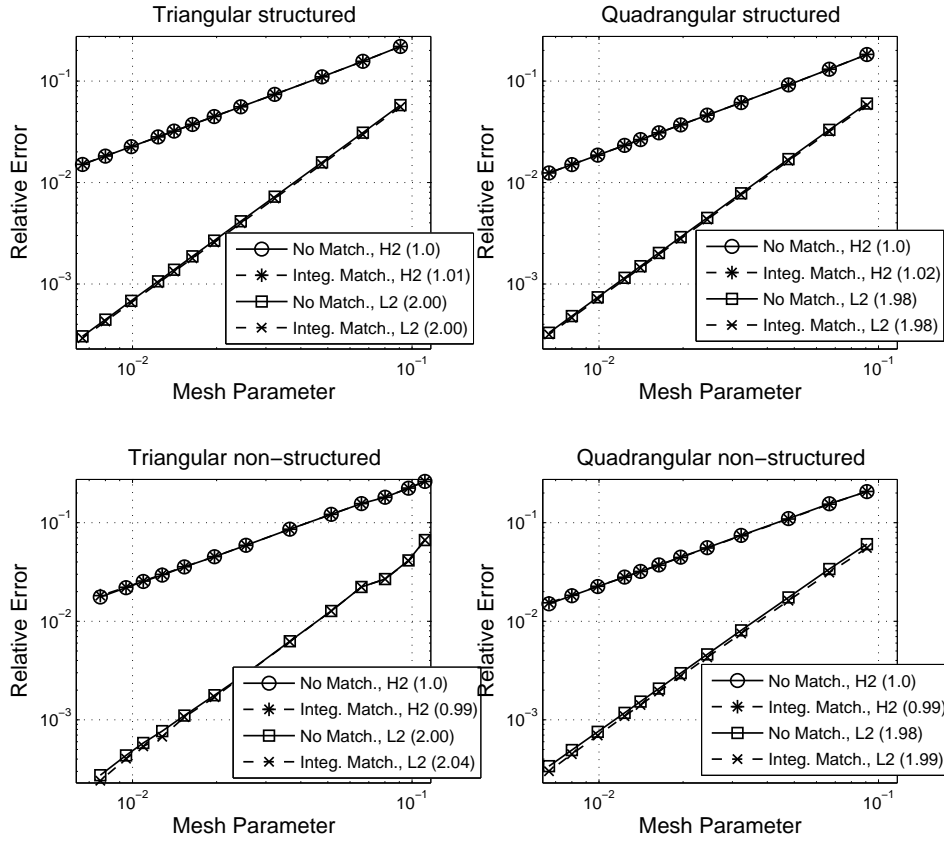


Figure 7: Comparison of the accuracy between classical FEM and integral matching.

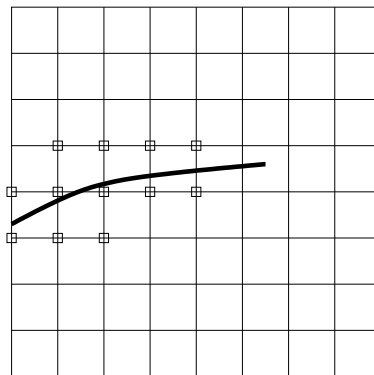


Figure 8: Set of nodes to be enriched along the crack.

is independent of the distance between nodes and crack: whether they are far or not, the non-invertibility remains. Let us also remark that it is not particularly due to the fact that the mesh is structured or not, even if, in our tests, the problem arises only with non-structured meshes. By the way, in the case of a straight crack and structured meshes, the quadrangles had always three sub-triangles crossed by the crack (see Fig. 10, right), which does not cause any non-invertibility.

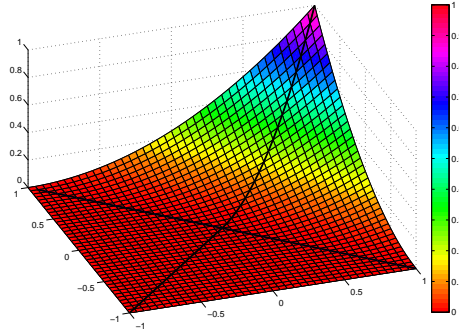


Figure 9: A particular FVS function: both this function and its derivatives are completely zero on two adjacent sub-triangles.

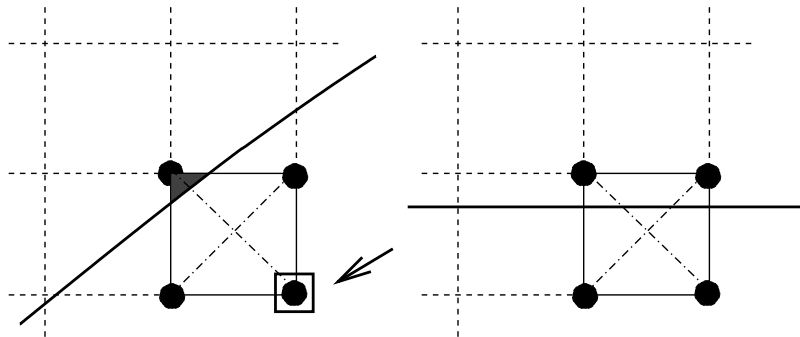


Figure 10: Left: only two sub-triangles are crossed, which leads to a singular matrix. Right: three subtriangles are crossed, which leads to an invertible matrix.

To solve this problem, several ideas were studied, like moving the crack or the nodes. However, they are not fully convenient. Moving the crack in order to avoid to cross two sub-triangles is quite complicated and, moreover, it may introduce a significant error. If we choose to move the nodes, it is mandatory to move almost all the nodes belonging to cracked quadrangles and to line up them with the crack. This method can be considered as a remeshing of a part of the domain. It is numerically expensive and not in good agreement with the philosophy of XFEM. So we tried something else. Starting from the fact that this incompatibility problem does not occur with the HCT triangle, the idea is then to divide each quadrangle crossed by the crack in two triangles (see an example in Fig. 12). It is an easy and quite cheap method, and we emphasize on the fact that it is not a remeshing: the nodes of the "new" mesh are exactly the same as those of the "old" one.

To conclude this section, let us notice that it may certainly exist other possibilities to overcome this non-invertibility. This could be an area for further research.

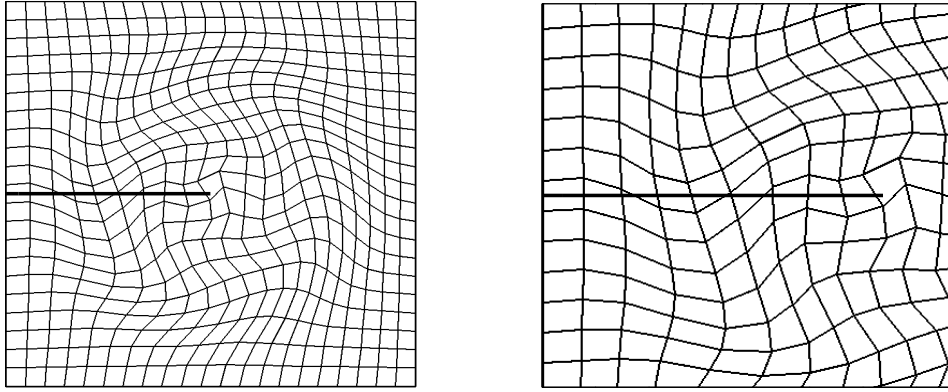


Figure 11: Left: example of mesh leading to a non-invertible linear system, with the crack (in bold line). Right: zoom on the crack.

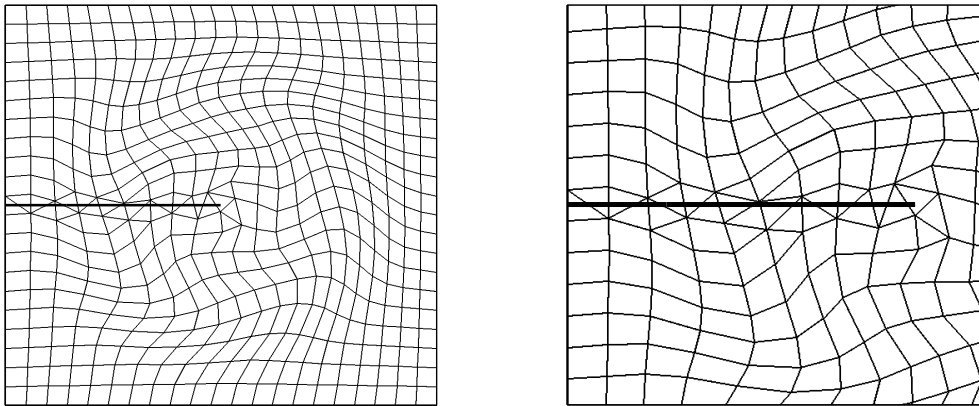


Figure 12: Left: entire modified mesh (the straight line is the crack). Right: zoom on the crack.

Remark 3 Let us specify that we also tested the possibility to remove the degree of freedom in the square, Fig. 10. We found it leads to a significant loss in accuracy, which prevent to obtain the optimal rate of convergence.

4.3 First enrichment strategy: adding degrees of freedom on each node

This strategy is directly inspired from [5, 4] for plane elasticity problems. Some optimal *a priori* error estimates are given in [26]. The main idea is that every node included in a fixed area will be enriched with the nonsmooth functions (see Fig. 13). This enrichment is obtained by multiplying the nonsmooth functions with the finite element shape ones of the reduced HCT/FVS element (both nodal and derivatives values), denoted φ_i . Contrary to [4], the shape functions are the same both in enriched and non-enriched terms. Then, the unknown reads

$$u^h = \sum_i a_i \varphi_i + \sum_{j \in J} b_j H \varphi_j + \sum_{k \in K} \sum_{l=1}^4 c_{kl} F_l \varphi_k. \quad (8)$$

where the nonsmooth functions are linked to the expression of the bilaplacian asymptotic solution (5) and given by

$$\begin{cases} F_1(r, \theta) = r^{3/2} \sin \frac{3\theta}{2} ; & F_2(r, \theta) = r^{3/2} \sin \frac{\theta}{2} ; \\ F_3(r, \theta) = r^{3/2} \cos \frac{3\theta}{2} ; & F_4(r, \theta) = r^{3/2} \cos \frac{\theta}{2}, \end{cases} \quad (9)$$

in polar coordinates relatively to the crack tip. Moreover, K denotes the set of degrees of freedom (dof) associated with the nodes that are inside the enrichment zone: they are indicated by a small circle in Fig. 13. At each circled node, both reduced HCT and FVS elements have three dof, named for instance $\varphi_{i_1}, \varphi_{i_2}, \varphi_{i_3}$. For each of these dof, we add 4 non-smooth enrichment functions: it means that for φ_{i_1} we introduce $F_1\varphi_{i_1}, F_2\varphi_{i_1}, F_3\varphi_{i_1}, F_4\varphi_{i_1}$. The procedure is the same for φ_{i_2} and φ_{i_3} . So it makes 12 additional dof due to enrichment functions, on each node inside the enrichment area. Let us also remark that some nodes are enriched both with H and nonsmooth functions.

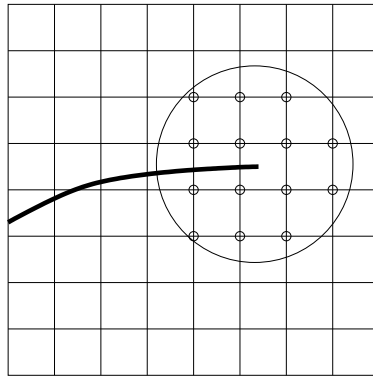


Figure 13: Set of nodes enriched by the nonsmooth functions.

4.4 Two dofs enrichment strategies

From [4], we know that a large number of additional degrees of freedom in the enrichment area generally leads to a very high condition number for the associated linear system. A first attempt in decreasing the number of nonsmooth enrichment dofs, and then decreasing the condition number, is obtained by making the following modification in the nonsmooth enrichment.

The asymptotic displacements near the crack (5) are function of the 2 unknown stress intensity factors K_1 and K_2 . But, in the XFEM enrichment (8), these functions are developed on 4 degrees of freedom. So an idea is to develop the asymptotic displacements on only 2 degrees of freedom. It means that the 4 F_l functions are replaced by the two following ones, say G_l

$$\begin{cases} G_1(r, \theta) = r^{3/2} \left(\frac{\nu + 7}{3(\nu - 1)} \cos \frac{3}{2}\theta + \cos \frac{\theta}{2} \right); \\ G_2(r, \theta) = r^{3/2} \left(\frac{3\nu + 5}{3(\nu - 1)} \sin \frac{3}{2}\theta + \sin \frac{\theta}{2} \right). \end{cases} \quad (10)$$

So the nonsmooth enrichment term $\sum_{k \in K} \sum_{l=1}^4 c_{kl} F_l \varphi_k$ becomes $\sum_{k \in K} \sum_{l=1}^2 c_{kl} G_l \varphi_k$ in the expression (8) of the approximate solution u^h . To conclude, let us just remark that this idea was tested in several references (for instance, in plane elasticity, see [27]).

4.5 Second enrichment strategy: global nonsmooth functions and integral matching

The second strategy is inspired by the so called "XFEM dof gathering with pointwise matching", introduced in [4]. The idea of such method is to reduce the number of degrees of freedom that are introduced in the finite element basis. For this, the support of the nonsmooth added functions is the whole enrichment area (see Fig. 14) and they are not multiplied by the finite element basis functions. So, instead of twelve additional degrees of freedom per node inside the enrichment area, there are only four nonsmooth degrees of freedom for the whole system: they are associated with the functions given by (9).

We introduce the boundary of the enrichment area, say Γ . It cuts Ω into two sub-domains: the enrichment area, say Ω_1 , and the rest of Ω , say Ω_2 (see Fig. 14). The unknowns defined on each domain Ω_i are denoted by u_i^h and their expressions read

$$\begin{cases} u_1^h = \sum_{i \in N_1} a_i \varphi_i + \sum_{j \in J_1} b_j H \varphi_j + \sum_{i=1}^4 c_i F_i; \\ u_2^h = \sum_{i \in N_2} a_i \varphi_i + \sum_{j \in J_2} b_j H \varphi_j, \end{cases} \quad (11)$$

where N_1 and N_2 are the sets of degrees of freedom that are located in Ω_1 and Ω_2 ($N_1 \cap N_2$ is not empty: it corresponds to the set of nodes that are on the boundary Γ). In a same way, J_i are the sets of degrees of freedom of J that are located in Ω_i ($J_1 \cap J_2$ is not empty for the same reason). Naturally, a matching condition is needed at the interface between the enrichment area and the rest of the domain, in order to insure the continuity of the function and its derivatives. Instead of

the pointwise matching introduced in [4], we use the integral matching given by (6). Note that a mathematical analysis of the optimality of such a method in the framework of plane elasticity is provided in [28, 29].

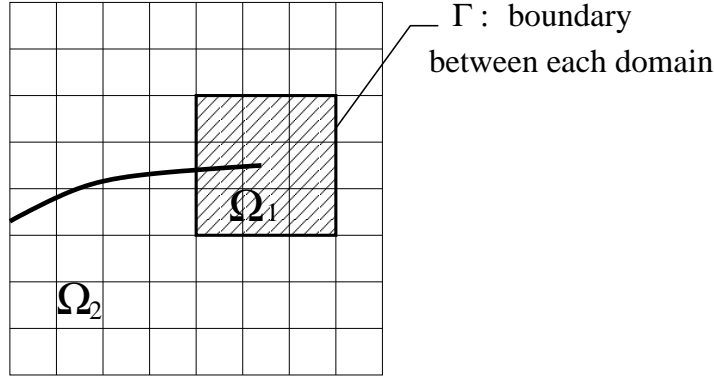


Figure 14: Set of elements which represents the support of the nonsmooth functions (set Ω_1).

Remark 4 *As in the previous section, it is also possible to develop the nonsmooth displacement on the two dofs given by (10). This possibility will be considered further.*

5 Numerical experiments

The numerical experiments presented in this section were performed with the open-source finite element library Getfem++ [23] developed by our team.

5.1 Test problems

We test the previous methods on two benchmark problems of a plate with a straight through crack. The plate is the unit square, and the material is assumed to be isotropic. In the experiments, the values of E and ν were taken at 14.98 kPa and 0.3 . Moreover, half the thickness ε is 0.045 . However, let us recall that, for the Kirchhoff-Love model, the thickness is irrelevant with respect to numerical accuracy.

First problem The exact solution is the pure mode II (see Fig. 15) which reads

$$u^{ex1}(r, \theta) = \frac{1}{10} r^{3/2} \left(\frac{3\nu + 5}{3(\nu - 1)} \sin \frac{3}{2}\theta + \sin \frac{\theta}{2} \right), \quad (12)$$

in polar coordinates. It is quite easy to check this function is such that $\frac{2E\varepsilon^3}{3(1-\nu^2)} \Delta^2 u = 0$, and satisfies free-edge boundary conditions on the crack faces.

Second problem The exact solution is the sum of the pure modes I and II, noted G_1 and G_2 on equation (10), and of a regular part. The expression reads :

$$u^{ex2} = G_1 + G_2 + 10 [x_1^4 - 6\nu x_1^2 x_2^2 + (2\nu - 1)x_2^4 + 2]. \quad (13)$$

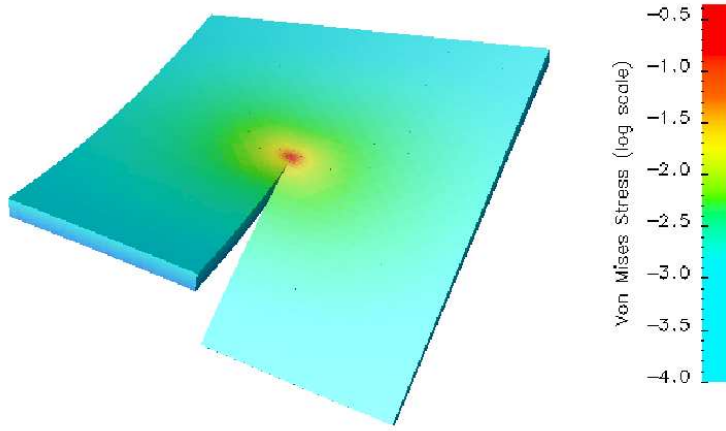


Figure 15: Graph of the first solution, with the Von Mises stress criteria.

The three terms G_1 , G_2 and regular part remaining are all such that $\frac{2E\varepsilon^3}{3(1-\nu^2)}\Delta^2 u = 0$, and satisfy the free-edge boundary conditions on the crack faces. The crack makes a $\pi/6$ angle with the line $\{x_2 = 0\}$.

For both these solutions (12)-(13), there is no volume nor surface forces: the loading is due to non-homogeneous boundary conditions, given by u^{ex_1} and u^{ex_2} .

5.2 Convergence Curves

The classical and eXtended Finite Element Methods (FEM and XFEM) were applied to the two previous problems. In particular, three XFEM strategies were investigated : they are called XFEM 1 (first strategy with four nonsmooth functions, see Section 4.3), XFEM 1 bis (first strategy with two nonsmooth functions, see Section 4.4) and XFEM 2 (second strategy, see Section 4.5). The L^2 (displacement) and H^2 (energy) errors were measured, on triangular, quadrangular, structured and non-structured meshes. Let us notice that the coarsest meshes had 11 elements on each edge of the domain, and the most refined ones 151. The enrichment area diameter was 0.30.

First test-case

The convergence curves are given in Fig. 16 and Fig. 17. They show the errors decrease when the mesh is refined. More precisely, we can first observe that the three XFEM are more accurate than classical FEM. Among them, the most accurate is the second enrichment strategy (XFEM 2) and the less one is the first strategy with two nonsmooth functions (XFEM 1 bis). So it exhibits that the use of four nonsmooth functions instead of the two asymptotics nonsmooth displacements leads to more accurate numerical results. A kind of superconvergence behaviour also appears for XFEM 1, especially with quadrangular meshes: thanks to it, its accuracy reaches XFEM 2 for the thinnest meshes. Moreover, it can be noticed that, for non-structured meshes and particularly for triangular ones, the displacement norm error is quite "oscillatory". However, we can see that the error globally decreases. So, even though this curve is less smooth than the others, the numerical results remain perfectible, but satisfactory.

Then, the rates of convergence were examined. Let us recall that, denoting u and u^h the analytical and numerical solutions, the so-called rate of convergence is the number α such that $\|u^h - u\|_V = O(h^\alpha)$. It also depends on the choice of space V . For example, on regular problems, with reduced HCT/FVS elements, we can expect $\|u^h - u\|_{H^2} = O(h)$ and $\|u^h - u\|_{L^2} = O(h^2)$ (see [22]). So, the results we obtained are summarized in Tab. 1. First, we observe that, for classical FEM, due to nonsmooth displacement at the crack tip, the rates of convergence remain limited around 0.5 in energy norm and 1 in displacement norm. Second, as expected, the XFEM rates of convergence are close to be optimal, *i.e.* they are the same as those of a regular problem. An exception is XFEM 1 which appears to be superconvergent.

	Struct. meshes		Non-Struct. meshes	
	H^2 norm	L^2 norm	H^2 norm	L^2 norm
FEM Tri.	0.51	0.95	0.50	0.94
FEM Quad.	0.51	0.94	Not tested	
XFEM 1 bis Tri.	0.95	1.77	1.10	2.11
XFEM 1 bis Quad.	0.94	1.73	1.35	2.61
XFEM 1 Tri.	1.19	2.36	1.27	2.67
XFEM 1 Quad.	1.44	3.23	1.72	3.65
XFEM 2 Tri.	1.02	2.09	1.04	1.94
XFEM 2 Quad.	1.05	2.11	1.06	1.95

Table 1: Rates of convergence for FEM and XFEM.

Second test-case

Only the two main strategies, named XFEM 1 and XFEM 2, were tested. We did not perform a numerical comparison with a classical finite element method. However we compared them with XFEM without singular enrichment, which means the use of equation (7) without the last term (it gives similar results compared to classical finite element method). The convergence curves are given Fig. 18 and 19, and the rates of convergence Tab. 2. These results lead to the same conclusions than the first test case:

- XFEM 2 is more accurate than XFEM 1 (however, roughly equal on non-structured meshes).
- The poor rate of convergence due to the crack is improved by the use of singular enrichment. The optimal rates are reached by both enriched strategies.

5.3 Condition number

The application of a finite element method leads to solve a linear system which can be more or less well-conditioned. This section deals with this point in XFEM strategies framework. The numerical results of the first test-case are given in Fig. 20 and 21, the second test-cas Fig. 22.

The classical FEM is taken as a reference. It can be observed that its condition number grows at a rate close to 4 which is the expected rate due to the fact that we solve a fourth order boundary value problem. As expected, since XFEM uses additional nonsmooth shape functions, its condition number is higher. More precisely, XFEM 1 produces the most ill-conditioned linear systems.

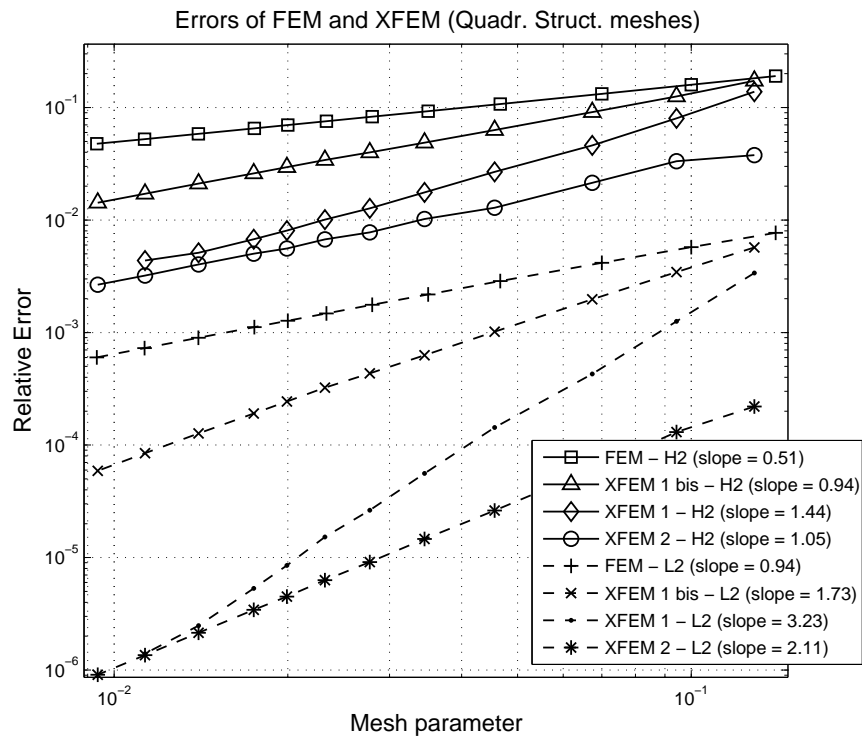
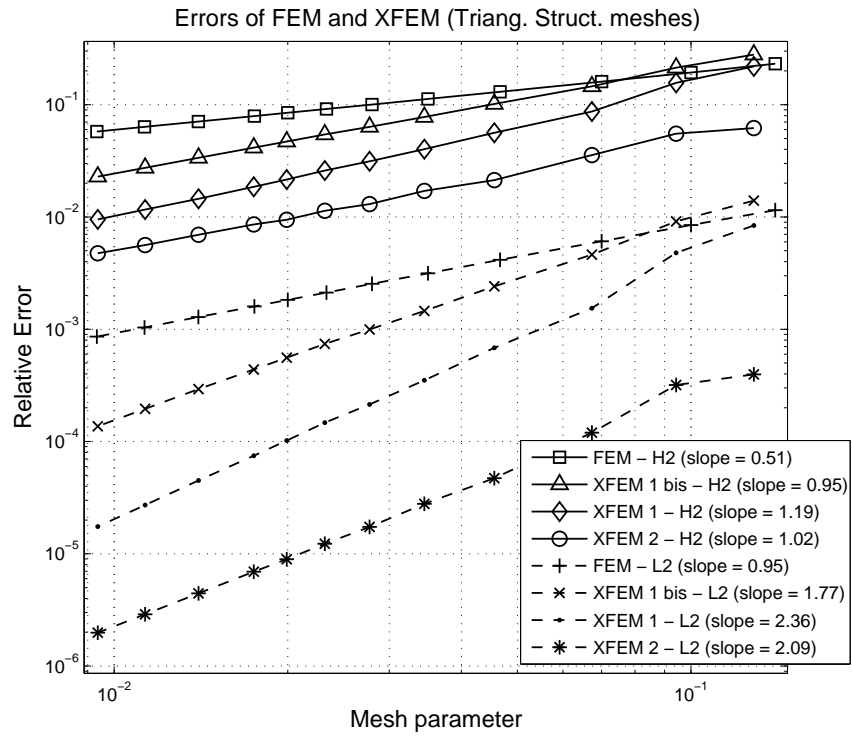


Figure 16: First test-case, convergence curves for FEM and XFEM on structured meshes. Top: triangular meshes. Bottom: quadrangular meshes.

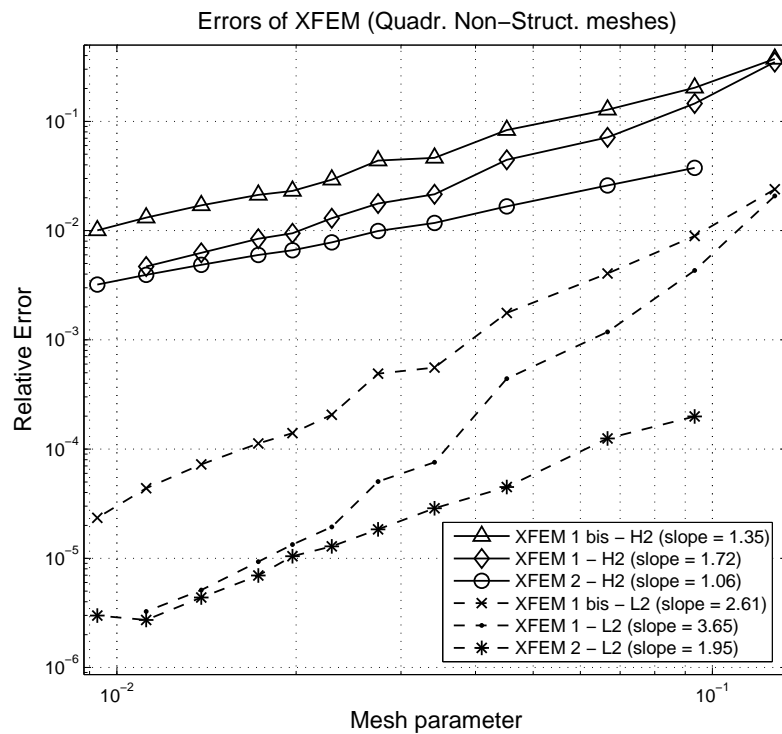
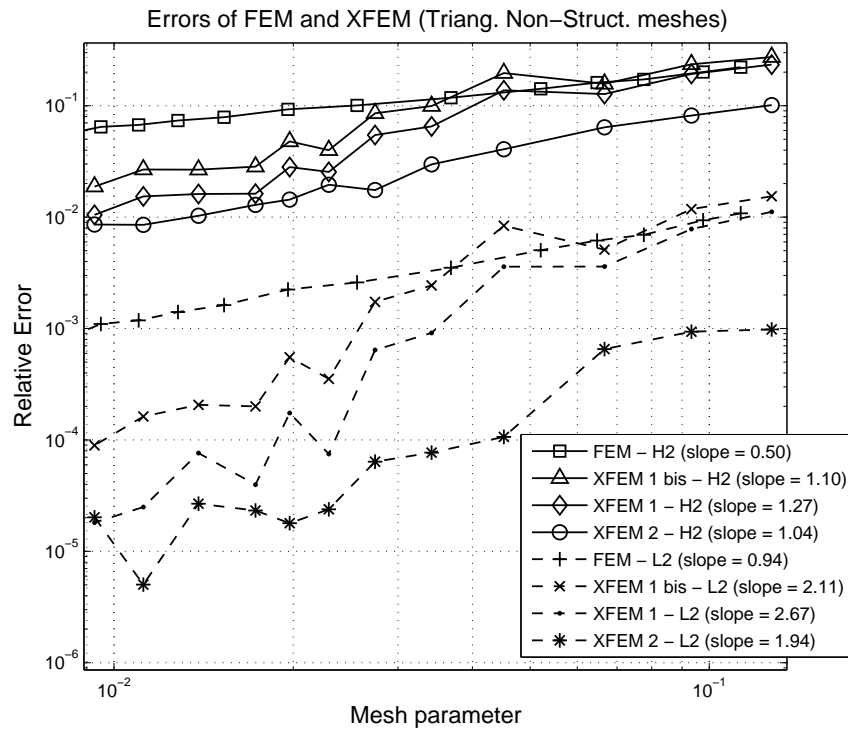


Figure 17: First test-case, convergence curves for FEM and XFEM on non-structured meshes. Top: triangular meshes. Bottom: quadrangular meshes.

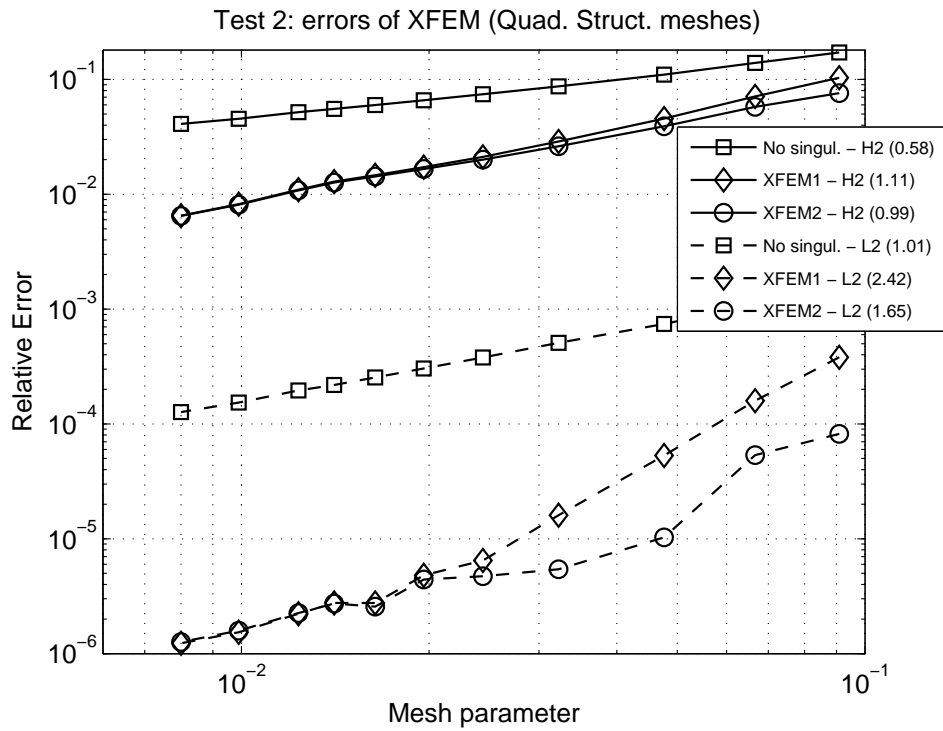
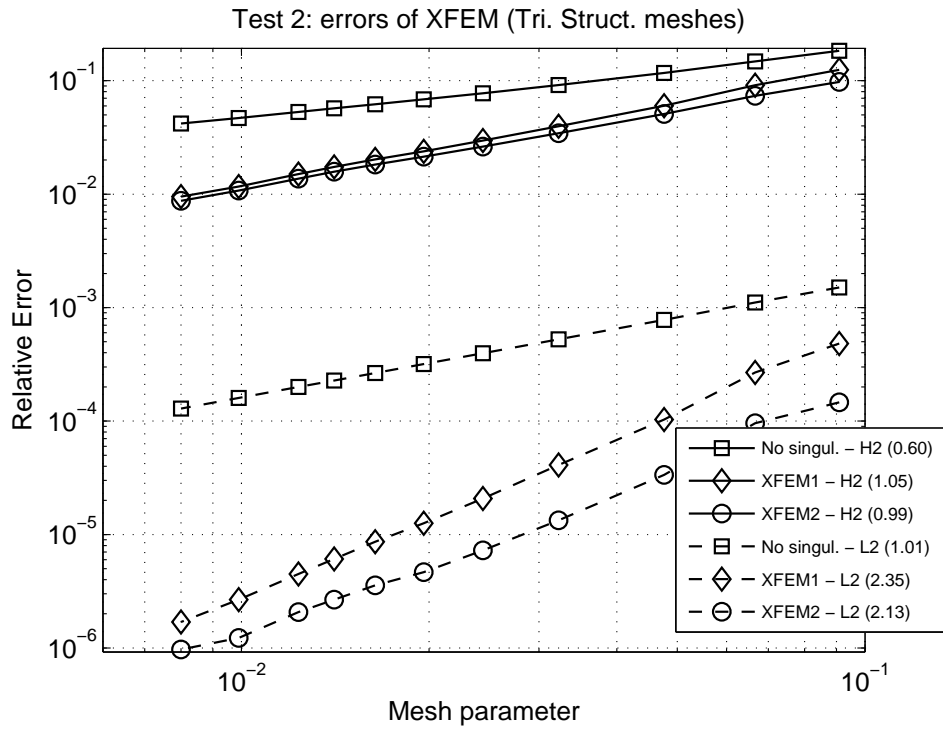


Figure 18: Second test-case, convergence curves for FEM and XFEM on structured meshes. Top: triangular meshes. Bottom: quadrangular meshes.

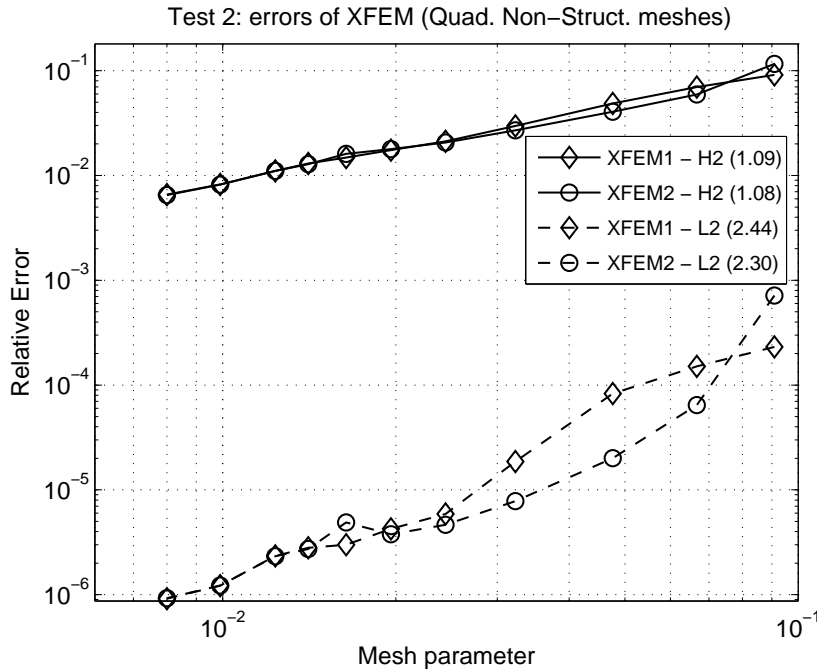
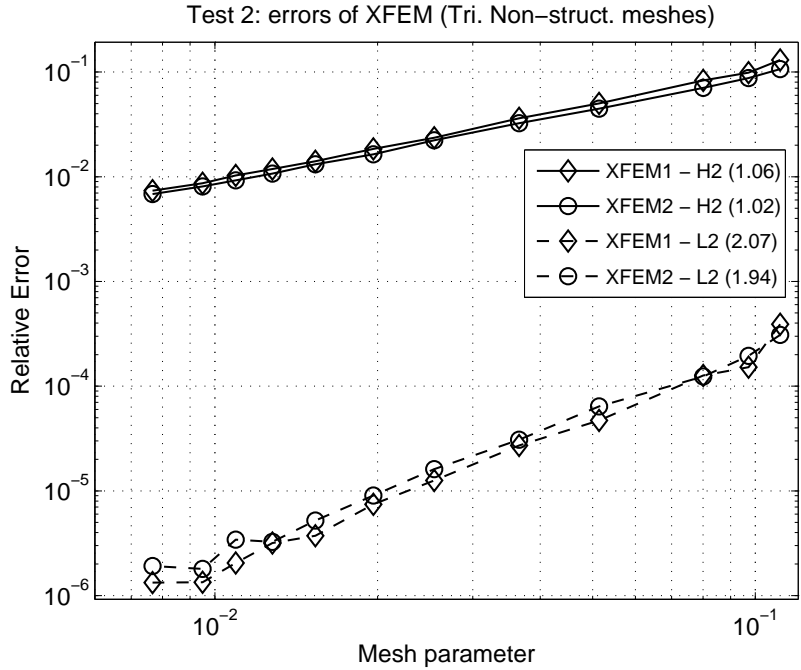


Figure 19: Second test-case, convergence curves for XFEM on non-structured meshes. Top: triangular meshes. Bottom: quadrangular meshes.

	Struct. meshes		Non-Struct. meshes	
	H^2 norm	L^2 norm	H^2 norm	L^2 norm
No. singul. Tri.	0.60	1.01	Not tested	
No. singul. Quad.	0.58	1.01	Not tested	
XFEM 1 Tri.	1.05	2.35	1.06	2.07
XFEM 1 Quad.	1.11	2.42	1.09	2.44
XFEM 2 Tri.	0.99	2.13	1.02	1.94
XFEM 2 Quad.	0.99	1.65	1.08	2.30

Table 2: Rates of convergence for FEM and XFEM.

It is close to 10^{16} for the thinnest meshes. So this phenomenon could limit the capabilities of such a method as, beyond this value, a matrix is close to be non-invertible at the machine precision. Moreover, the rates of growth are the hugest: between 6.5 and 8.4. XFEM 2 and XFEM 1 bis produce better conditioned linear systems since their rate of growth remains close to 4, like the classical FEM.

Let us recall that the high condition number of XFEM 1 was already mentionned in [4, 5]. So, to avoid this drawback, alternative enrichment strategies were developed. In [5], the authors introduced a numerical procedure that allows, in a sense, to orthogonalize the enrichment functions. Some key features of our "Integral Matching" strategy (XFEM 2) are directly inspired from [4]: the use of global enrichment functions instead of enriching each node decreases the number of dofs and also the condition number. Hence, it is not surprising to observe that the condition number of XFEM 2 is significantly lower than XFEM 1. However, what can be observed here is that the development of the nonsmooth displacement on two dofs, instead of four, is also a very efficient way to reduce the condition number, even if it leads to a slight loss of accuracy.

Remark 5 *We also tested the nonsmooth displacement developed on two dofs for the Integral Matching enrichment strategy. The nonsmooth enrichment term $\sum_{k=1}^4 c_k F_k$ of (11) is simply replaced by $\sum_{k=1}^2 c_k G_k$, where functions G_k are expressed in (10). With this modification, the errors remain unchanged: their numerical values are exactly the same on about 3 significant figures. Hence the curves are the same as those of XFEM 2 in Fig. 16 and 17. It is why we do not present any graphic here. The condition number is just slightly decreased compared to XFEM 2: it is divided by a factor that varies between 1 and 3 for most of the meshes.*

Concluding remarks

We presented an adaptation of XFEM to the Kirchhoff-Love plate model. Several XFEM formulations have been investigated and we stress the fact that the features of these methods make possible an implementation in an industrial software:

- for all formulations, it was shown that the accuracy was nearly equal to the one of a classical finite element method on a regular non-cracked problem;

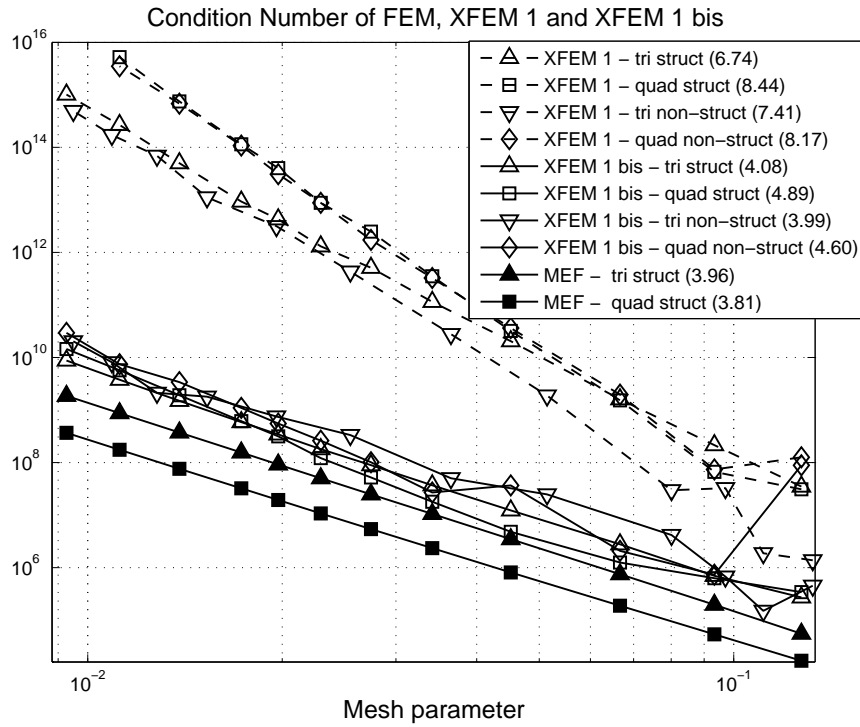


Figure 20: First test-case, condition numbers of FEM and XFEM first strategy (with 2 or 4 nonsmooth dofs).

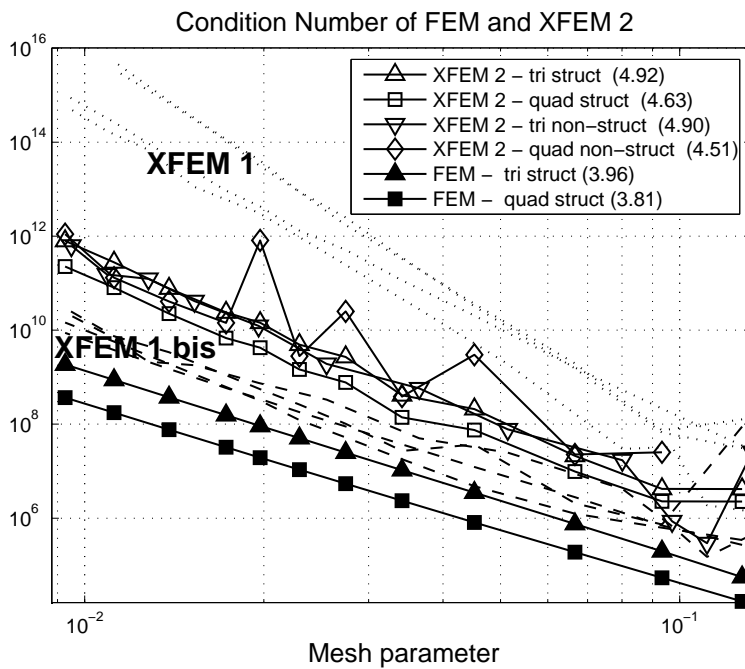


Figure 21: First test-case, condition numbers of FEM and all XFEM.

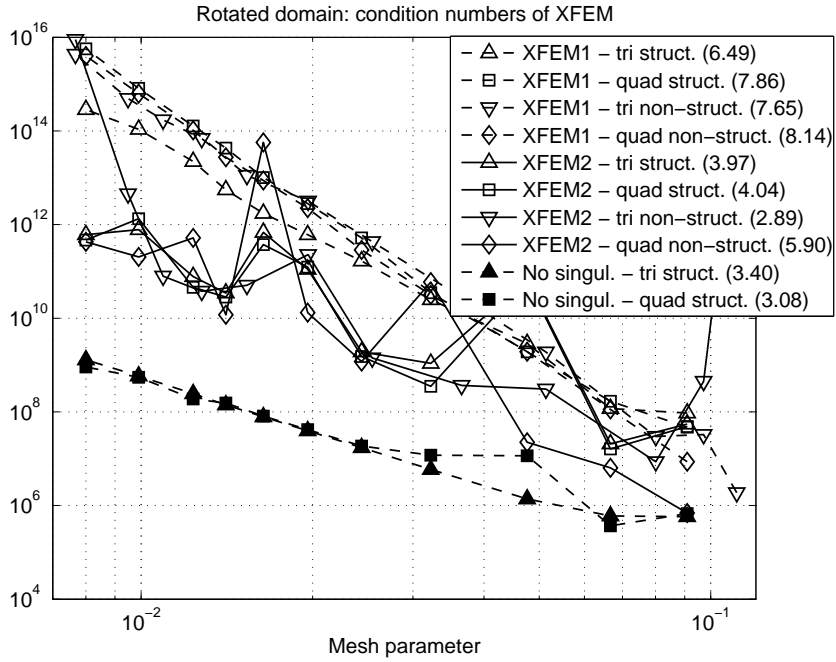


Figure 22: Second test-case, condition numbers of all XFEM (with/without singular enrichment).

- the computational cost is nearly equal to the one of the Mindlin-Reissner model, which is the reference model for plate problems in industrial softwares;
- our formulation is not subject to numerical locking due to the enrichment, unlike the Mindlin-Reissner model and the strategy presented in [12].

The numerical experiments suggest that the use of the integral matching formulation is the best choice in terms of accuracy (especially on coarse meshes) and in terms of condition number of the linear system.

Finally, the computation of stress intensity factors, which is one of the key applications of the numerical methods on cracked domains, will be presented in a forthcoming paper.

Acknowledgments

This work was supported by a joint contract with Airbus (n° 001847) and "Centre National de la Recherche Scientifique" (n° 060131), and also supported by "Agence Nationale de la Recherche" (project ANR-05-JCJC-0182-01). The authors wish to thank Marc Balzano (Airbus France) and Patrick Laborde (University of Toulouse, France) for their confidence and support all along this work.

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