# Stress Intensity Factors computation for bending plates with XFEM

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## SUMMARY

The modelization of bending plates with through the thickness cracks is investigated. We consider the Kirchhoff-Love plate model which is valid for very thin plates. Reduced Hsieh-Clough-Tocher triangles and reduced Fraejis de Veubeke-Sanders quadrilaterals are used for the numerical discretization. We apply the eXtended Finite Element Method (XFEM) strategy: enrichment of the finite element space with the asymptotic bending singularities and with the discontinuity across the crack. The main point, adressed in this paper, is the numerical computation of stress intensity factors. For this, two strategies, direct estimate and J-integral, are described and tested. Some practical rules, dealing with the choice of some numerical parameters, are underlined. Copyright © 2010 John Wiley & Sons, Ltd.

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## 1. INTRODUCTION

In the framework of linear elastic fracture mechanics, the computation of Stress Intensity Factors is one of the most important problems. Although some analytical solutions can be found in literature, they always correspond to simple geometries and loads. For general geometry and loading conditions, numerical methods have to be employed. So the goal of this paper is to investigate efficient numerical tools for very thin cracked plates, such as those which are widely used for instance in aircraft structures. Let us also remark only through the thickness cracks will be considered here and that the material the plate is made of is homogeneous and isotropic.

So the first tool we use is XFEM (eXtended Finite Element Method). This is a strategy initially developed for plane elasticity cracked problems (see [1, 2]) and it is now the subject of a wide literature (among many others, see [3, 4, 5, 6, 7, 8, 9] and references therein). It mainly consists in the introduction of the discontinuity across the crack and of the asymptotic displacements into the finite element space.

At the moment, there are few previous works devoted to the adaptation of XFEM to plate or shell models. In [10, 11, 12], shell models are used: since the near tip asymptotic displacement in this

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model is unknown, no near-tip enrichment is used but only the discontinuous one. In particular, in [12, 13], the crack tip is always on an element edge: it means the crack span entire elements of the mesh. Moreover, the Mindlin-Reissner is used and the crack propagation is investigated, which is not the case of our work. In [14], which deals with cracked shells, the cracked part of the domain is modelized by a three-dimensional XFEM formulation. It is matched with the rest of the domain, formulated with a classical finite element shell model. In this paper, a plate model is kept on the entire domain, and we consider singular enrichment. In [15], the plate model used is the Mindlin-Reissner one. However, in this reference, an important locking effect for thin plates has been detected despite the use of some classical locking-free elements. This suggests that this locking effect is due to the XFEM enrichment.

Even though most of the finite element codes are based on the Mindlin-Reissner plate model, the so-called Kirchhoff-Love model provides also a realistic description of the displacement for a thin plate since it is the limit model of the three-dimensional elasticity model when the thickness vanishes (see [16]). For instance, the panels used in aeronautic structures can be about a few millimeters thin, for several meters long. On this kind of plates, the shear effect can generally be neglected and consequently the Kirchhoff-Love model is mechanically appropriate. It has already been used for the purpose of fracture mechanics (for instance, see [17]). Moreover, for through-thickness cracks, the limit of the energy release rate of the three-dimensional model can be expressed with the Kirchhoff-Love model solution (see [18] and [19]).

Since the Kirchhoff-Love model corresponds to a fourth order partial differential equation, a conformal finite element method needs the use of  $C^1$  (continuously differentiable) elements. We consider the reduced Hsieh-Clough-Tocher triangle (reduced HCT) and the reduced Fraejis de Veubeke-Sanders quadrilateral (reduced FVS) because they are the less costly conformal  $C^1$  elements [20]. In the XFEM framework, the knowledge of the asymptotic crack tip displacement is required. It is the case for a Kirchhoff-Love isotropic plate as it corresponds to the bilaplacian singularities (see [21]). Thanks to all this material, it was possible to derive an efficient XFEM for thin cracked plates with Kirchhoff-Love theory. It is detailed in [22] and some of its features, used in this paper, are recalled in the following.

Then, it is possible to introduce the second tool which deals with Stress Intensity Factors (SIF) computation. For this, two different strategies are suggested. The first one consists in a direct estimate. It follows an idea introduced in [23], for two-dimensional elasticity problem. And we have adapted it to our bending plate problem and our XFEM formulation. Let us remark this approach lies on the SIF definition as the limit, when the distance, say r, to the crack tip tends to 0, of some stress multiplied by  $\sqrt{r}$ , up to a mutiplicative coefficient. In classical FEM, the stress is numerically evaluated and strongly depends on the mesh refinement. In [12] for plates, an alternative approach is suggested, which uses the knowledge of the asymptotic singular displacements and the numerical evaluation of the displacements through what is called a displacement extrapolation technique. It will be explained later we can directly use the singular displacements we have introduced in the numerical formulation. The second is more classical and uses J-integral. This is the way chosen in [15] for Mindlin-Reissner plates. In our case, we had to derive this approach for Kirchhoff-Love model.

The paper is organized as follows. Section 2 describes the model problem. Section 3 is devoted to the extended finite element discretization of the Kirchhoff-Love model. In Section 4, two strategies for SIF computation are detailed. In the last section, numerical results are presented and discussed, which illustrate the capabilities of these methods, and enable to derive some practical rules for the choice of some numerical parameters.

## 2. THE MODEL PROBLEM

#### 2.1. Notations and variational formulation

Let us consider a thin plate, *i.e.* a plane structure for which one dimension, called the thickness, is very small compared to the others. For this kind of structures, starting from *a priori* hypotheses



Figure 1. The cracked thin plate (the thickness is oversized for the sake of clarity).

on the expression of the displacement fields, a two-dimensional problem is usually derived from the three-dimensional elasticity formulation by means of integration along the thickness. Then, the unknown variables are set down on the mid-plane of the plate, denoted here by  $\omega$ .

This mid-plane  $\omega$  is an open subset of  $\mathbb{R}^2$ . In the three-dimensional cartesian referential, the plate (see Fig. 1) occupies space

$$\left\{ \left(x_1, x_2, x_3\right) \in \mathbb{R}^3 \mid (x_1, x_2) \in \omega \text{ and } x_3 \in \left[ -\varepsilon; \varepsilon \right[ \right\} \right\}$$

So, the  $x_3$  coordinate corresponds to the transverse direction, and all the mid-plane points have their third coordinate equal to 0. The thickness is  $2\varepsilon$ . Finally, we assume that the plate has a through the thickness crack and that the material is homogeneous and isotropic, of Young's modulus E and Poisson's ratio  $\nu$ .

In plate theory, the following approximation of the three-dimensional displacements is usually considered

$$\begin{cases} u_1(x_1, x_2, x_3) &= \underline{u}_1(x_1, x_2) + x_3 \phi_1(x_1, x_2) , \\ u_2(x_1, x_2, x_3) &= \underline{u}_2(x_1, x_2) + x_3 \phi_2(x_1, x_2) , \\ u_3(x_1, x_2, x_3) &= u_3(x_1, x_2) . \end{cases}$$
(1)

In these expressions,  $\underline{u}_1$  and  $\underline{u}_2$  are the membrane displacements of the mid-plane points while  $u_3$  is the deflection,  $\phi_1$  and  $\phi_2$  are the section rotations. In the case of an isotropic material, the variational formulation splits into two independent problems: the first, called the membrane problem, deals only with membrane displacements, while the second, called the bending problem, concerns deflection and rotations. The membrane problem corresponds to the classical plane elasticity problem and has been already treated in many references (see for instance [4, 5]). So, here, we only consider the bending problem.

In industrial finite element codes, the most widely used plate model is the Mindlin-Reissner one, for which the displacement is given by (1). Nevertheless, for reasons mentioned in the introduction, we choose here to work with the Kirchhoff-Love model, which can be seen as a particular case of (1), as it is obtained by introducing the so-called Kirchhoff-Love assumptions, which read

$$\nabla u_3 + \phi = 0$$
 i.e.  $\begin{cases} \phi_1 = -\partial_1 u_3 \\ \phi_2 = -\partial_2 u_3 \end{cases}$  (2)

where the notation  $\partial_{\alpha}$  stands for the partial derivative with respect to  $x_{\alpha}$ . A first consequence of this relation is that the transverse shear strain is identically zero, which avoids the shear locking problem. A second consequence of (2) is that the section rotation only depends on the transverse displacement. It means that this displacement is the only unknown function for the bending problem. For convenience, it will be denoted by u in the following. So, in the Kirchhoff-Love framework and



Figure 2. Fracture modes for Kirchhoff-Love bending model ( $\Gamma_C$  is the crack). Left: a symmetric bending leads to mode I. Right: a shear bending leads to mode II.

for a pure bending problem, the three-dimensional displacement reads

$$\begin{cases} u_1(x_1, x_2, x_3) &= -x_3 \partial_1 u(x_1, x_2) \\ u_2(x_1, x_2, x_3) &= -x_3 \partial_2 u(x_1, x_2) \\ u_3(x_1, x_2, x_3) &= u(x_1, x_2) \end{cases},$$

For the sake of simplicity, we assume the plate is clamped on its boundary and the crack faces are traction free. Then, the plate is subjected to a volume force, say f of coordinates  $(f_1, f_2, f_3)$ , and two surface forces, say  $g^+$  and  $g^-$ , applied on the top and bottom surfaces. The variational formulation (or virtual work formulation) of the Kirchhoff-Love model reads as

$$\begin{cases} \text{Find } u \in H_0^2(\omega) \text{ such that for any } v \in H_0^2(\omega) \\ \int_{\omega} \frac{2E\varepsilon^3}{3(1-\nu^2)} \left[ (1-\nu) \,\partial_{\alpha\beta}^2 u + \nu \,\Delta u \,\delta_{\alpha\beta} \right] \,\partial_{\alpha\beta}^2 v \,dx = \int_{\omega} \left[ F \,v - M_\alpha \,\partial_\alpha v \right] \,dx. \end{cases}$$
(3)

where:

$$-F = \int_{-\varepsilon}^{\varepsilon} f_3 dx_3 + g_3^+ + g_3^-, \text{ which is the resulting transverse loading,}$$
$$-M_{\alpha} = \int_{-\varepsilon}^{\varepsilon} x_3 f_{\alpha} dx_3 + \varepsilon (g_{\alpha}^+ - g_{\alpha}^-), \text{ which is the resulting moment loading.}$$

Moreover  $\delta_{\alpha\beta}$  stands for the Kronecker's symbol and the summation convention over repeated indices is adopted, Greek indices varying in  $\{1, 2\}$ . Finally,  $H_0^2(\omega)$  is the classical Sobolev space of square integrable functions whose first and second derivatives in the distributions sense are square integrable, and which vanish on the boundary, like their normal derivative (see [24] for instance).

## 2.2. Asymptotic displacement near the crack tip and fracture modes

In the Kirchhoff-Love plate model, there are two fracture modes. Applying a symmetric bending leads to the first fracture one, while applying an anti-symmetric bending or a transverse shear, leads to the second (see Fig. 2).

To characterize them, let us recall that the governing equation related to the bending variational problem (3) reads

$$\frac{2E\varepsilon^3}{3(1-\nu^2)}\Delta^2 u = F + \partial_\alpha M_\alpha \quad , \tag{4}$$

on the mid-plane  $\omega$ . It is a bilaplacian problem for which the singularities are well-known (see [21]). So, close to the crack tip, the displacement may be written as  $u = u_r + u_s$ , where  $u_r$  stands for the regular part of the transverse displacement and belongs to  $H^3(\omega)$ . The singular part  $u_s$  reads

$$u_{s}(r,\theta) = \mathcal{A}_{KL} r^{3/2} \left[ K_{1} \left( \frac{\nu+7}{3(\nu-1)} \cos \frac{3}{2}\theta + \cos \frac{\theta}{2} \right) + K_{2} \left( \frac{3\nu+5}{3(\nu-1)} \sin \frac{3}{2}\theta + \sin \frac{\theta}{2} \right) \right]$$
(5)

in polar coordinates relatively to the crack tip (see Fig. 3), with

$$\mathcal{A}_{KL} = \frac{\sqrt{2}}{2} \frac{1 - \nu^2}{E\varepsilon(3 + \nu)} \quad . \tag{6}$$

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Figure 3. System of polar coordinates relatively to the crack tip (the crack is in bold line).

This singular displacement belongs to  $H^{5/2-\eta}(\omega)$  for any  $\eta > 0$ . In addition, the scalar coefficients  $K_1$  and  $K_2$  are the so-called "Stress Intensity Factors", which are widely used in fracture mechanics for crack propagation.

To conclude, we recall that the Kirchhoff-Love plate theory corresponds to the limit of the threedimensional elasticity theory, when the thickness vanishes. However, the singularities we present here are deduced from the Kirchhoff-Love theory, and not from the three-dimensional elasticity theory. The reader interested by the link between the singularities of these two theories is referred to [25].

## 3. EXTENDED FINITE ELEMENT APPROXIMATION OF THE KIRCHHOFF-LOVE MODEL

## 3.1. Classical finite element approximation

Let us introduce now the finite element discretization of the variational formulation (3). In order to have a conformal method, the finite element space  $V^h$  has to satisfy  $V^h \subset H_0^2(\omega)$ . This leads to the use of  $C^1$  finite elements. Among the available elements having this regularity, the reduced HCT triangles (see [20], p. 356-357) and FVS quadrangles (see [20], p. 359-360) are of particular interest. For both elements, the triangle (resp. quadrangle) is divided into three (resp. four) sub-triangles (see Fig. 4). The basis functions are  $P_3$  polynomials on each sub-triangle and matched  $C^1$  across each internal edge. In addition, to decrease the number of dof (degrees of freedom), the normal derivative is assumed to vary linearly along the external edges of the elements (this assumption does not hold on the internal edges). At the end, there are only three dof on each node for both elements: the value of the function and its first derivatives. So, these elements have the two following advantages:

- 1. The computational cost is limited to three dof for each node of the mesh, like a classical Mindlin-Reissner element (the deflection and the two section rotations).
- 2. The theoretical error is in O(h) and  $O(h^2)$  for the  $H^2$  and  $L^2$  norm (respectively), on regular problems (*h* stands for the mesh parameter). The minimum regularity assumption for this error estimate to hold is that the exact solution belongs to  $H^3(\omega)$  (see [26]).

So, the reduced HCT or FVS elements and standard Mindlin elements have the same features as far as numerical cost and accuracy are concerned.

#### 3.2. XFEM enrichment

To define our XFEM enrichment strategy, we follow ideas presented in previous papers [4, 5]. As usual, the discontinuity of the displacement across the crack is represented using Heaviside-like function, which is multiplied by the finite element shape functions. For the nonsmooth enrichment close to the crack tip, an enrichment area of fixed size is defined and the nonsmooth functions are added inside all this area. The strategy, which is used in this paper, is inspired by the so called "XFEM dof gathering with pointwise matching", introduced in [4] and developed for plates in [22].

As noticed in [9], such functions enrichment scheme does not satisfy a local partition of unity since enriched basis functions do not vanish at the edges of enriched elements. To solve this problem,



Figure 4. HCT triangle and FVS quadrangle. Location of degrees of freedom and sub-triangles.

the authors introduced the so-called "shifted" Heaviside function, which is of particular interest when the so-called branch functions, which reads  $\sqrt{r}\cos(\theta/2)$  in their case, is not used for the enrichment, as they did in their paper. However, here, we use the branch functions. So we do not use the "shifted" Heaviside function because, first, optimal convergence results for our finite element scheme were already obtained numerically (see [22]), and, second, Nicaise & al [27] have theoretically proved that an approach such as [4] is optimal.

Let us now describe more precisely the enrichment. So, let  $\Gamma$  be the boundary of the enrichment area (see Fig. 5). It cuts  $\omega$  into two sub-domains: the enrichment area, say  $\omega_1$ , and the rest of the domain, say  $\omega_2$ . Then, the support of the singular added functions is the whole enrichment area but they are not multiplied by the finite element basis functions. So, instead of 6 additional dof per node inside the enrichment area, there are only 2 singular dof for the whole system. Consequently, if the unknowns defined on each sub-domain  $\omega_i$  are denoted by  $u_i^h$ , their expressions read

$$\begin{cases}
 u_1^h = \sum_{i \in N_1} a_i \varphi_i + \sum_{i \in J_1} b_i H \varphi_i + \sum_{i=1}^2 c_i F_i , \\
 u_2^h = \sum_{i \in N_2} a_i \varphi_i + \sum_{i \in J_2} b_i H \varphi_i ,
\end{cases}$$
(7)

where  $\varphi_i$  are the basis functions of the reduced HCT/FVS elements. The jump of H function is located on the crack; the set J denotes the dof whose shape function support is completely crossed by the crack (see Fig. 6). Furthemore,  $N_1$  and  $N_2$  are the set of dof that are located in  $\omega_1$  and  $\omega_2$  $(N_1 \cap N_2$  is not empty and corresponds to the set of nodes that are on the boundary  $\Gamma$ ). In a same way,  $J_i$  are the set of dof of J that are located in  $\omega_i$  and  $J_1 \cap J_2$  is not empty for the same reason. Finally, the singular enrichment functions, derived from (5), are

$$\begin{cases} F_1 &= r^{3/2} \left( \frac{\nu + 7}{3(\nu - 1)} \cos \frac{3}{2}\theta + \cos \frac{\theta}{2} \right) &, \\ F_2 &= r^{3/2} \left( \frac{3\nu + 5}{3(\nu - 1)} \sin \frac{3}{2}\theta + \sin \frac{\theta}{2} \right) &. \end{cases}$$

Naturally, a matching condition is needed at the interface between the enrichment area and the rest of the domain, in order to insure the continuity of the function and its derivatives. The following relations were chosen at this aim

$$\begin{cases} \int_{\Gamma} u_1^h \lambda &= \int_{\Gamma} u_2^h \lambda , \quad \forall \lambda \in \Lambda , \\ \int_{\Gamma} \partial_n u_1^h \mu &= \int_{\Gamma} -\partial_n u_2^h \mu , \quad \forall \mu \in M , \end{cases}$$

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Figure 5. Set of elements which represents the support of the nonsmooth functions.

where  $\Lambda$  and M are appropriate multiplier spaces. Here,  $\Lambda$  is the space of piecewise polynomials of degree 2 and M piecewise polynomials of degree 1, and we have checked in [22] that this choice keeps an optimal rate of convergence for the finite element scheme. Finally, let us observe the change of sign in front of the normal derivative  $\partial_n u_i$  is due to the fact that the outside normal vector has an opposite sign whether it is used in  $\partial_n u_1^h$  or  $\partial_n u_2^h$ .



Figure 6. Set of nodes to be enriched along the crack.

## 4. COMPUTATION OF STRESS INTENSITY FACTORS

In industrial applications dealing with cracked structures, the plate displacement is not straightforwardly meaningful in terms of crack propagation. The SIF are linked to the energy release rate G, and they provide such an information (we have  $K_1^2 + K_2^2$  proportional to G). However, the calculation of SIF usually needs the use of some specific post-treatments, such as computation of J-integral for instance.

## 4.1. First method: direct estimate

An interesting feature of the previously described methodology is that it can lead to a direct estimate of SIF. Comparing expressions of the asymptotic displacement (5) with the numerical displacement (7), it appears that, if the method is convergent, the finite element coefficients  $(c_i)_i$  should be close to  $(K_i)_i$ , up to a multiplicative constant, we shall calculate now.

Actually, in the expression of the singular displacement of Kirchhoff-Love theory (5), it appears two singular modes. However, in the above XFEM formulation, the singular enrichments  $F_1$  and  $F_2$ 

are exactly these two singular fonctions. In particular, in the sub-domain  $\omega_1$  containing the crack tip, the numerical solution reads

$$u_1^h = \sum_{i \in N_1} a_i \varphi_i + \sum_{i \in J_1} b_i H \varphi_i + \sum_{i=1}^2 c_i F_i .$$

To show how coefficients  $c_i$  can be good approximations of SIF, up to a multiplicative constant to be determined, let us go back to the mathematical definitions of the SIF, in Kirchhoff-Love theory, which are

$$K_{1} = \lim_{r \to 0} \sqrt{2r} \sigma_{22}(r, \theta = 0, x_{3} = \varepsilon) ,$$
  

$$K_{2} = \lim_{r \to 0} \frac{3 + \nu}{1 + \nu} \sqrt{2r} \sigma_{12}(r, \theta = 0, x_{3} = \varepsilon) .$$
(8)

The singular stresses are in  $O(1/\sqrt{r})$  in the vicinity of the crack tip. However, if we calculate the components  $\sigma_{12}$  and  $\sigma_{22}$  resulting from numerical displacement  $u_1^h$ , multiply the result by  $\sqrt{r}$  and make r tends to 0, all the regular terms are cancelled and only the coefficients  $c_i$  remain, up to a multiplicative constant. So these coefficients fit well with the SIF definitions. We only have to evaluate the multiplicative constant.

Now, let us give the calculation in details for  $K_1$ , the same procedure being convenient for  $K_2$ . Under the assumption of isotropic and homogeneous material, we recall the link between  $\sigma_{22}$  and u

$$\sigma_{22} = -x_3 \frac{E}{1-\nu^2} \left[\nu \partial_{11}^2 u + \partial_{22}^2 u\right]$$

Replacing u by  $u_1^h$  in this expression and reporting it in (8) leads to

$$K_1^h = -\frac{E \varepsilon \sqrt{2}}{1-\nu^2} \left( \nu \underbrace{\lim_{r \to 0} \sqrt{r} \partial_{11}^2 u_1^h}_{l_1} + \underbrace{\lim_{r \to 0} \sqrt{r} \partial_{22}^2 u_1^h}_{l_2} \right)$$

These two limits  $l_1$  and  $l_2$  exist. Since the most singular part of  $u_1^h$  is in  $O(r^{3/2})$ , we have  $\partial_{\alpha\beta}^2 u = O(r^{-1/2})$ . Apart from the crack, the element edges and the internal boundaries of the HCT/FVS elements, the basis functions of  $u_1^h$  are  $C^2$ , so  $\partial_{\alpha\beta}^2 u_1^h$  exists, and we have

$$\lim_{r \to 0} \sqrt{r} \,\partial_{\alpha\beta}^2 u_1^h = \lim_{r \to 0} \sum_i c_i \sqrt{r} \,\partial_{\alpha\beta}^2 F_i \quad ,$$

as

$$\lim_{r \to 0} \sqrt{r} \, \partial^2_{\alpha\beta} \varphi_i \; = \; 0 \quad ; \quad \lim_{r \to 0} \sqrt{r} \, \partial^2_{\alpha\beta} \varphi_i \; H \; = \; 0 \quad .$$

The calculation of the second derivatives of  $F_i$  fonctions is not difficult and gives

$$\lim_{r \to 0} \sqrt{r} \,\partial_{11}^2 F_1(r,0) = \frac{\nu+1}{\nu-1} , \quad \lim_{r \to 0} \sqrt{r} \,\partial_{22}^2 F_1(r,0) = \frac{\nu-3}{\nu-1}$$
$$\lim_{r \to 0} \sqrt{r} \,\partial_{11}^2 F_2(r,0) = 0 , \quad \lim_{r \to 0} \sqrt{r} \,\partial_{22}^2 F_2(r,0) = 0 .$$

These expressions show that  $F_2$  is not involved in the estimation of  $K_1^h$ . We deduce that  $l_1 = c_1 \frac{\nu + 1}{\nu - 1}$  and  $l_2 = c_1 \frac{\nu - 3}{\nu - 1}$ , and finally

$$K_1^h = - \frac{\sqrt{2} E \varepsilon (3+\nu)}{1-\nu^2} c_1$$

The calculation for  $K_2$  can be carried out the same way. The definition (8) involves  $\sigma_{12}$ , which is proportional to  $\partial_{12}^2 u$ . So, here, we have to calculate the cross derivatives of functions  $F_i$  and obtain

$$\lim_{r \to 0} \sqrt{r} \,\partial_{12}^2 F_1(r,0) = 0 \quad , \quad \lim_{r \to 0} \sqrt{r} \,\partial_{12}^2 F_2(r,0) = \frac{\nu+1}{\nu-1} \quad ,$$

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which gives

$$K_2^h = - \frac{\sqrt{2} E \varepsilon (3+\nu)}{1-\nu^2} c_2 .$$

As the numerical values of coefficients  $c_i$  result directly from the solving of the linear system associated to the calculation of  $u_1^h$ , no post-treatment is necessary to obtain approximations  $K_1^h$  and  $K_2^h$  of the SIF.

To conclude this section, let us remark that a similar idea has already been described and tested in [23]. In this paper, a numerical method, close to the one we propose here, is applied on a two-dimensional elasticity problem. This method uses the XFEM formulation named "geometrical enrichment" in [5] and "XFEM with fixed enrichment area" in [4], except that an enrichment zone of fixed area is not defined. The authors prefer to select from one to three layers of nodes surrounding the crack tip. Three meshes are used, the mesh parameter being divided by two at each refinement. With a single layer of enriched nodes, the error on the SIF is around 15% and the mesh refinement does not improve significantly the accuracy. Adding a second layer of enriched nodes makes the error fall to globally 1% and, with the third layer, under 1%. However, this paper shows clearly that the mesh refinement does not lead to a strict decrease of the error. Finally, let us mention a recent work of Nicaise & al [27], which shows a rather slow theoretical convergence of order  $\sqrt{h}$ for bi-dimensional elasticity with XFEM.

## 4.2. Second method: J-integral computation

*4.2.1. Method description and formulation* For Kirchhoff-Love theory, the expression of the J-integral has already been established in [17]. Its expression is

$$J = -\frac{1}{2} \int_{\Gamma} m_{\alpha\beta} \,\partial_{\alpha\beta} u \,b_1 \,dl + \int_{\Gamma} m_{\alpha\beta} \,b_\beta \,\partial_{1\alpha} u \,dl - \int_{\Gamma} \partial_{\alpha} \,m_{\alpha\beta} \,b_\beta \,\partial_1 u \,dl \quad,$$

where  $m_{\alpha\beta}$  stands for the bending moment and  $b_{\alpha}$  for the outward unit vector normal to the contour of integration  $\Gamma$ . However, this expression is not the one used in numerical computations, since it does not allow to separate the contributions of each SIF in the energy release rate. In addition, it needs to carry out integrations on contours, which is not well suited for finite element computations. The usual technique allowing to do accurate SIF calculations via J-integrale is described in [1]. It is based upon works of Destuynder, described in details for instance in [19].

Now, the formulation adapted to the case of Kirchhoff-Love plate theory is presented. It follows quite closely the one described in [1], which deals with the case of two-dimensional elasticity. So, J-integral can be rewritten

$$J = \int_{\Gamma} m_{\alpha\beta} \left( \partial_{1\alpha} u \, b_{\beta} \, - \, \frac{1}{2} \, \partial_{\alpha\beta} u \, b_1 \right) \, dl \, - \, \int_{\Gamma} \partial_{\alpha} \, m_{\alpha\beta} \, b_{\beta} \, \partial_1 u \, dl \quad .$$

Following [1], we introduce two states. State (1)  $(m_{\alpha\beta}^{(1)}, u^{(1)})$  matches the numerical solution which SIF we want to evaluate. State (2)  $(m_{\alpha\beta}^{(2)}, u^{(2)})$  is an auxiliary state corresponding to the asymptotic displacement of mode I or II, depending on the SIF we want to calculate. The J-integral for the sum of these two states reads

$$J^{(1+2)} = \int_{\Gamma} \left( m^{(1)}_{\alpha\beta} + m^{(2)}_{\alpha\beta} \right) \left[ \left( \partial_{1\alpha} u^{(1)} + \partial_{1\alpha} u^{(2)} \right) b_{\beta} - \frac{1}{2} \left( \partial_{\alpha\beta} u^{(1)} + \partial_{\alpha\beta} u^{(2)} \right) b_{1} \right] dl - \int_{\Gamma} \left( \partial_{\alpha} m^{(1)}_{\alpha\beta} + \partial_{\alpha} m^{(2)}_{\alpha\beta} \right) \left( \partial_{1} u^{(1)} + \partial_{1} u^{(2)} \right) b_{\beta} dl .$$

It is developed as

$$J^{(1+2)} = J^{(1)} + J^{(2)} + I^{(1,2)} , (9)$$

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where  $I^{(1,2)}$  is the so-called interaction integral

$$I^{(1,2)} = \int_{\Gamma} \left( m^{(1)}_{\alpha\beta} \partial_{1\alpha} u^{(2)} + m^{(2)}_{\alpha\beta} \partial_{1\alpha} u^{(1)} \right) b_{\beta} - \frac{1}{2} \left( m^{(1)}_{\alpha\beta} \partial_{\alpha\beta} u^{(2)} + m^{(2)}_{\alpha\beta} \partial_{\alpha\beta} u^{(1)} \right) b_{1} dl - \int_{\Gamma} \left( \partial_{\alpha} m^{(1)}_{\alpha\beta} \partial_{1} u^{(2)} + \partial_{\alpha} m^{(2)}_{\alpha\beta} \partial_{1} u^{(1)} \right) b_{\beta} dl \quad .$$
(10)

Introducing now the formula, established in [17], which links J-integral to SIF

$$J = \frac{2\varepsilon\pi(1+\nu)}{3E(3+\nu)} \left(K_1^2 + K_2^2\right)$$

we rewrite it in the case of the sum of the 2 states and find

$$J^{(1+2)} = J^{(1)} + J^{(2)} + 2 \frac{2\varepsilon\pi(1+\nu)}{3E(3+\nu)} \left(K_1^{(1)}K_1^{(2)} + K_2^{(1)}K_2^{(2)}\right) \quad . \tag{11}$$

Since the right hand sides of (9) and (11) are equal, we deduce

$$I^{(1,2)} = \frac{4\varepsilon\pi(1+\nu)}{3E(3+\nu)} \left(K_1^{(1)}K_1^{(2)} + K_2^{(1)}K_2^{(2)}\right)$$

So if, in this relation, state (2) is mode I (with  $K_1^{(2)} = 1$  et  $K_2^{(2)} = 0$ ), the value of the SIF  $K_1$  is obtained with the value of the interaction integral, since previous equation becomes

$$I^{(1,2)} = \frac{4\varepsilon\pi(1+\nu)}{3E(3+\nu)} K_1^{(1)} \quad .$$
(12)

 $K_2$  can be calculated in the same way.

4.2.2. *Transformation of the interaction integral into a domain integral* The previous section shows that calculating interaction integral (10) with singular crack fields enables to deduce the values of the SIF with (12). However, for numerical purpose, the interaction integral is transformed into a domain integral. Here again, we follow [1].

First, let us rewrite the interaction integral (10) in a more compact form

$$I^{(1,2)} = \int_{\Gamma} (\mathcal{A}_{\beta} b_{\beta} + \mathcal{B} b_1) \ dl \ ,$$

with

$$\begin{aligned} \mathcal{A}_{\beta} &= \left( m_{\alpha\beta}^{(1)} \partial_{1\alpha} u^{(2)} + m_{\alpha\beta}^{(2)} \partial_{1\alpha} u^{(1)} \right) - \left( \partial_{\alpha} m_{\alpha\beta}^{(1)} \partial_{1} u^{(2)} + \partial_{\alpha} m_{\alpha\beta}^{(2)} \partial_{1} u^{(1)} \right) \\ \mathcal{B} &= -\frac{1}{2} \left( m_{\alpha\beta}^{(1)} \partial_{\alpha\beta} u^{(2)} + m_{\alpha\beta}^{(2)} \partial_{\alpha\beta} u^{(1)} \right) . \end{aligned}$$

The value of  $I^{(1,2)}$  remains unchanged if the integrand is multiplied by a regular function, say q, whose value is 1 on the area defined by  $\Gamma$ , et 0 on another contour  $C_0$  that encloses  $\Gamma$ . So, if we assume there is no surface force applied on disc A defined by contour  $C_0$ ,  $I^{(1,2)}$  reads also

$$I^{(1,2)} = \int_{\Gamma} \left( \mathcal{A}_{\beta} B_{\beta} + \mathcal{B} B_{1} \right) q \, dl \quad .$$

where C is defined by  $C = \Gamma \cup C_+ \cup C_- \cup C_0$ , while B denotes the outward normal to C (see Fig. 7). Then, using divergence theorem and taking the limit of contour  $\Gamma$ , when  $\Gamma$  tends to the point (0, 0), the contour integral becomes a surface one and domain A becomes the complete disc that contains the crack tip and which is bounded by  $C_0$ . Thus, we have

$$I^{(1,2)} = \int_{A} \left[ \partial_{\beta} \left( \mathcal{A}_{\beta} q \right) + \partial_{1} \left( \mathcal{B} q \right) \right] dA$$
  
= 
$$\int_{A} \left[ \left( \partial_{\beta} \mathcal{A}_{\beta} + \partial_{1} \mathcal{B} \right) q + \mathcal{A}_{\beta} \partial_{\beta} q + \mathcal{B} \partial_{1} q \right] dA$$

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Figure 7. Integration contours for  $I^{(1,2)}$  calculation.

A direct calculation shows easily that  $\partial_{\beta} \mathcal{A}_{\beta} + \partial_{1} \mathcal{B} = 0$ . Hence, we obtain

$$I^{(1,2)} = \int_{A} \left[ \left( m^{(1)}_{\alpha\beta} \partial_{1\alpha} u^{(2)} + m^{(2)}_{\alpha\beta} \partial_{1\alpha} u^{(1)} \right) - \left( \partial_{\alpha} m^{(1)}_{\alpha\beta} \partial_{1} u^{(2)} + \partial_{\alpha} m^{(2)}_{\alpha\beta} \partial_{1} u^{(1)} \right) \right] \partial_{\beta} q \, dA$$
$$- \frac{1}{2} \int_{A} \left( m^{(1)}_{\alpha\beta} \partial_{\alpha\beta} u^{(2)} + m^{(2)}_{\alpha\beta} \partial_{\alpha\beta} u^{(1)} \right) \partial_{1} q \, dA \quad .$$

Finally, setting  $D = \frac{2E\varepsilon^3}{3(1-\nu^2)}$ , let us observe that

$$m_{\alpha\beta}^{(1)} \partial_{\alpha\beta} u^{(2)} = -D \left[ (1-\nu) \partial_{\alpha\beta}^2 u^{(1)} + \nu \Delta u^{(1)} \delta_{\alpha\beta} \right] \partial_{\alpha\beta}^2 u^{(2)}$$
  
=  $-D \left[ (1-\nu) \partial_{\alpha\beta}^2 u^{(1)} \partial_{\alpha\beta}^2 u^{(2)} + \nu \Delta u^{(1)} \Delta u^{(2)} \right] = m_{\alpha\beta}^{(2)} \partial_{\alpha\beta} u^{(1)}$ 

Hence, the final expression of interaction integral reads

$$I^{(1,2)} = \int_{A} \left[ \left( m_{\alpha\beta}^{(1)} \partial_{1\alpha} u^{(2)} + m_{\alpha\beta}^{(2)} \partial_{1\alpha} u^{(1)} \right) - \left( \partial_{\alpha} m_{\alpha\beta}^{(1)} \partial_{1} u^{(2)} + \partial_{\alpha} m_{\alpha\beta}^{(2)} \partial_{1} u^{(1)} \right) \right] \partial_{\beta} q \, dA$$
$$- \int_{A} m_{\alpha\beta}^{(1)} \partial_{\alpha\beta} u^{(2)} \partial_{1} q \, dA \quad . \tag{13}$$

4.2.3. Numerical calculation of the interaction integral Now, our purpose is to calculate the interaction integral  $I^{(1,2)}$  given by (13), in the the case of Kirchhoff-Love model, treated with reduced HCT/FVS elements. Expression (13) contains three terms. There is no difficulty for the two ones which contain the bending moments  $m_{\alpha\beta}^{(i)}$  without derivatives. But the third term, which includes  $\partial_{\alpha}m_{\alpha\beta}^{(i)}$  is harder to handle, as it involves third order derivatives of the displacements. On the one hand, the functions we integrate are surely not in  $H^3(\Omega)$ . On the other hand, it cannot be expected that the third derivatives of a function may be correctly approximated by reduced HCT/FVS elements: for these elements, error estimates are only obtained up to the second derivatives. So we will transform (13) in order to avoid these third derivatives.

The expression, we want to modify, reads

$$X = - \int_{A} \left( \partial_{\alpha} m_{\alpha\beta}^{(1)} \partial_{1} u^{(2)} + \partial_{\alpha} m_{\alpha\beta}^{(2)} \partial_{1} u^{(1)} \right) \partial_{\beta} q \, dA$$

It is split in two terms

$$X = \underbrace{-\int_{A} \partial_{\alpha} m_{\alpha\beta}^{(1)} \partial_{1} u^{(2)} \partial_{\beta} q \, dA}_{X_{1}} \underbrace{-\int_{A} \partial_{\alpha} m_{\alpha\beta}^{(2)} \partial_{1} u^{(1)} \partial_{\beta} q \, dA}_{X_{2}} .$$

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 $X_2$  can be computed without any particular difficulty, as it depends only on the crack tip singular functions. It only needs computation of third derivatives of these singularities.  $X_1$  is integrated by parts

$$X_1 = \underbrace{\int_A m_{\alpha\beta}^{(1)} \partial_\alpha (\partial_1 u^{(2)} \partial_\beta q) \, dA}_{X_{11}} - \underbrace{\int_{\partial A} m_{\alpha\beta}^{(1)} \partial_1 u^{(2)} \partial_\beta q \, b_\alpha \, dl}_{X_{12}}$$

There is no problem concerning  $X_{11}$ . As far as  $X_{12}$  is concerned, in the case where  $u^{(2)}$  is the exact mode I, it can be checked that  $\partial_1 u^{(2)}$  cancels along the crack (this term cancels when  $\theta = \pi$ ), and then  $X_{12} = 0$ . But in the case of mode II,  $X_{12}$  calculation is more difficult: this term differs from 0, but only along the crack where  $\partial_{\beta}q$  is not 0. It is along the intersection between the crack and the boundary of the ring of integration. Nevertheless, in our numerical tests, for the mode II, we shall neglect this term. Despite this simplification, computations of  $K_2$  were not less precise than those of  $K_1$ .

To conclude this section, let us present briefly some features of numerical implementation. The calculation of interaction integral  $I^{(1,2)}$  needs to define explicitly function q. Let us recall this function is identically equal to 1 inside an area containing the crack tip, 0 outside a zone enclosing the first one, and q matches regularly from one zone to the other. Since only derivatives of q are needed in (13), those functions differ from 0 on a ring between the two zones. In practice, we define a ring of elements around the crack tip on which the J-integral is evaluated (see Fig. 8). In our numerical tests, this ring is made of elements located at a certain distance  $\mathcal{R}_J$  from the crack tip. Furthemore, the function q is represented on the reduced HCT/FVS basis. The nodal values are set to 1 on the internal boundary of the ring, to 0 on the external boundary while the degrees of freedom associated to the derivatives are set to 0 on both boundaries.



Figure 8. Ring of elements enclosing the crack tip.

## 5. NUMERICAL RESULTS

## 5.1. Description of the numerical study

The numerical experiments presented in this section were performed with the open-source finite element library Getfem++ [28].

5.1.1. Test cases Two test-cases with a straight through crack are considered in this paper. The solution of the first one is the sum of the two singular modes

$$u^{ex} = F_1 + F_2$$
.

The sides of the crack follow a free edge condition. On the rest of the domain boundary, a nonhomogeneous Dirichlet condition is given, whose value corresponds to  $u^{ex}$ . Consequently, the exact values of  $K_1$  and  $K_2$  are  $1/\mathcal{A}_{KL}$ , where  $\mathcal{A}_{KL}$  is defined by (6). Finally, the plate we took is the square  $[-0.5, 0.5] \times [-0.5, 0.5]$ , with the crack tip at the origin.

The second test case is more classical and comes from [29]. It consists in a square plate with a central straight through crack of length 2a, and a constant moment  $M_0$  is applied on the edges parallel to the crack. The dimensions of the plate are said to be "infinite", which means that reference SIF values are correct only if the crack is small compared to the dimensions of the plate. These reference values are

$$K_1 = \frac{3 M_0 \sqrt{a}}{2 \varepsilon^2} ; \quad K_2 = 0$$

For the numerical tests, we took a plate of edge 1, with a crack of size 2a = 0.2. This remains significant compared to [30], where calculations are carried out with 2a = 0.18. Since the problem is symmetric, only half of the domain is considered.



Figure 9. Second test case. Plate with central crack subjected to moments applied on two edges.

5.1.2. Goals of the study The aim of the numerical experiments is to study the error made by our SIF calculation methods, with respect to the following parameters:

- mesh parameter h,
- enrichment radius  $\mathcal{R}$  which corresponds to the "size" of  $\omega_1$  (see Figure 5),
- integration ring radius  $\mathcal{R}_J$ , for J-integral method only,
- structured or non-structured meshes.

In addition, for J-integral, results are compared with non-enriched Finite Element Method.

Another goal of these numerical experiments is to bring elements of answer to the question of the influence of parameters  $\mathcal{R}$  and  $\mathcal{R}_J$  and to propose eventually some practical rules for the choice of these parameters, depending on the mesh size h. Indeed, in [5] and [4], the enrichment area is a disc, of radii 0.05 and 0.1, respectively. In [22], we took  $\mathcal{R} = 0.15$ . However, in a more general manner, we think the choice of  $\mathcal{R}$  depends on the result we try to set. For example, to show the convergence of an enriched finite element method in  $L^2$  or  $H^2$  norm, taking a fixed value independant of h is convenient. Nevertheless, on the most refined meshes, the choice of fixed  $\mathcal{R}$ leads to enrich numerous layers of elements, which may be not necessary if we use only one mesh. So, in our study, we introduce two strategies for the choice of the size  $\mathcal{R}$  of enriched domain  $\omega_1$ . First, we consider several fixed values of  $\mathcal{R}$ . Second,  $\mathcal{R}$  depends on h, in such a way the enrichment area covers several layers of elements around the crack tip. It means  $\mathcal{R}$  is equal to k h, where k is an integer we have taken between 1 and 5. Let us remark also that taking  $\mathcal{R} = h$  is very close to the first XFEM formulation [1, 15], where only the element contening the crack tip is enriched by singular functions. Finally, as the results with the fixed value of  $\mathcal{R}$  were not more accurate than those with  $\mathcal{R}$  depending on h, we only present results with  $\mathcal{R} = k h$  in this paper. For more details, the reader is referred to [31].

## 5.2. Direct Estimate

This first method was tested on the two above mentioned test-cases, with triangular and quadrangular, structured and non-structured meshes, for several values of the mesh parameter h. Moreover, we have tested  $\mathcal{R} = k h$ , where k goes from 1 to 5.

The results for the first test case, are given Fig. 10 and Fig. 11. They show the method provides very good estimates of SIF. The relative error is always lower than 5% and often lower than 1%. Let us remark that an error of 5% is precise enough for many industrial applications. Nevertheless, the convergence can be very slow on non-structured meshes. Maybe it is due to high conditionning of the method, which reaches  $10^{12}$  on such meshes.

For the second test case, the size of the crack is a = 0.11 on half domain, which is the rectangle  $[0, 0.5] \times [-0.5, 0.5]$ . So the crack is smaller than in the first test case. Moreover, the enrichment area must not touch the boundary  $0 \times [-0.5, 0.5]$ , since it corresponds to a symmetry condition. Indeed, singular enrichment does not satisfy this condition. Here, we use meshes which the level of refinement is equivalent to those of the first test case. It leads to a more drastic constraint on the choice of  $\mathcal{R}$ . We also tested the same values of k, but a high value of k needs an initial level of refinement more important. For example, for k = 5, the less refined mesh, in structured quadrangular meshes, needs around 60 elements on the longest edge of the domain. This explains why some curves are not complete. However, when this level of refinement is reached, the error is lower than 5%. The results are presented Fig. 12.

Despite its slow convergence, the "direct estimate" method is simple, efficient, and provides SIF values close to the exact ones. According to the tests, increasing  $\mathcal{R}$  improves the results. So, due to the slow convergence, it may be more interesting to increase  $\mathcal{R}$  than to refine the mesh. We observe also that  $\mathcal{R} = 5$  h enables to reach always a satisfactory accuracy. It leads us to propose the following practical rule. Given a crack of length a, the domain has to be meshed with a minimum h around a/5 and the radius of the enrichment area is taken equal to 5 h. Let us remark this rule indicates that the smaller is the crack, the more the mesh has to be refined in ordre to take care of the crack. This is in accordance with intuition: the more a crack is small compared to elements size, the less it has influence on global solution. A very refined mesh is then necessary in order to "catch" its effect.

#### 5.3. J-integral

The same numerical experiments than in the previous section were carried out. But, here, the radius of the ring of integration  $\mathcal{R}_J$  has also been investigated.

5.3.1. First test case We have observed that, even if the results are accurate, from one mesh to another, the error is not strictly decreasing, as the value provided by J-integral oscillates around the exact value. Hence, a mesh can give an error slightly greater than a coarser one. That's why we give convergence curves only on the first test case, and on structured meshes, for which less oscillatory results are obtained.

So Fig 13 and Fig. 14 present convergence curves for structured meshes, both triangular and quadrangular. For this particular purpose, the radius of the enrichment area  $\mathcal{R}$  must be fixed, and it is equal to 0.15 here. The comparison with a non-enriched Finite Element Method (FEM) shows that XFEM improves the SIF values and that the rate of convergence may be slightly better.

Now, let us present a more global study, in which the numerical values of SIF are investigated, with respect to h,  $\mathcal{R}$  and  $\mathcal{R}_J$ . Fig. 15 gives results for  $\mathcal{R} = k h$  on non-structured triangular and quadrangular meshes. Moreover, only results on  $K_1$  were shown, curves for  $K_2$  being very similar



Figure 10. SIF calculations - Direct estimate - First test case - Triangular meshes.

[31]. For brevity, we do not present structured meshes results. In fact, they do not bring additionnal informations, and they have already been presented in the case of direct estimate (see Fig. 13 and 14).

So, our results show that the error often remains lower than 5%. On structured meshes, this error is generally less than 1% [31]. On non-structured meshes, taking  $\mathcal{R} = 3 h$  is enough to obtain an error lower than 5% on all meshes. Such a value for  $\mathcal{R}$  seems to be minimal. Besides, on coarser meshes, with  $\mathcal{R} = h$ , the error is often greater than 10%.

All in all, results are relatively stable with respect to ring radius  $\mathcal{R}_J$ . To conclude, it can be observed that Fig. 15 shows oscillations. Let us notice it is not the case for regular meshes [31]. That's why we explain it by the fact that, in our calculations, the ring of integration is only one element width, which may be too irregular on non-structured meshes to have stable results.



Figure 11. SIF calculations - Direct estimate - First test case - Quadrangular meshes.

Naturally, this explanation should be numerically tested. However, the error level on SIF appears to be good enough to avoid a more complex estimate.

5.3.2. Second test case We recall the crack is smaller here, which limits the choice of  $\mathcal{R}$  and  $\mathcal{R}_J$ . Again, we take  $\mathcal{R} = 3 h$  for k = 1, ..., 5. In all cases, the radius of the ring of integration  $\mathcal{R}_J$  varies from 0.05 to 0.11, so that this ring can touch the boundary. Our results tend to show the precision depends mainly on  $\mathcal{R}_J$ . When  $\mathcal{R}_J$  increases, the approximate SIF is closer to the exact one and the best values are obtained for the greatest. Finally, except coarser meshes, the best value is always lower than 5%, while the meshes with less than 2 elements on the crack induces significant errors. Numerical results are brought together Fig. 16, for non-structured meshes.



Figure 12. SIF calculations - Direct estimate - Second test case.

To conclude on this second test case, we observe that an as great as possible ring of integration must be chosen, in order to have the most accurate SIF. Then, the rule of construction, we propose, is still to take h = a/5 (for a crack of length a) and  $\mathcal{R} = 5 h$ .

## 6. CONCLUDING REMARKS

This paper adresses the modelization of bending plates with through the thickness cracks in the framework of linear elastic fracture mechanics. As very thin plates are considered, the Kirchhoff-Love plate model is used. The main point, studied in this paper, is the numerical computation of SIF. For that purpose, two strategies are described and evaluated on two test cases.



Figure 13. SIF convergence curves - J-integral - First test case - Triangular meshes.



Figure 14. SIF convergence curves - J-integral - First test case - Quadrangular meshes.

First, the "direct estimate" method is simple, efficient, and provides SIF values close to the exact ones. According to the tests, increasing the radius  $\mathcal{R}$  of the enrichment area improves the results. Moreover, it seems more interesting to increase  $\mathcal{R}$  than to refine the mesh. Second, a "J-integral" approach is derived which gives also good results. Furthemore, the comparison with a classical Finite Element Method shows that XFEM improves the SIF values.

Finally, a practical rule may be emphasised. In all our tests, a radius  $\mathcal{R} = 5 h$  enables to reach always a satisfactory accuracy, for both SIF computation strategies. To make it possible, it leads to the following mesh rule. Given a crack of length a, the domain has to be meshed with a minimum h around a/5 and the radius of the enrichment area will be taken equal to 5 h.

Naturally, some developments and applications of this work have to be done. The first one deals with crack propagation as in [7, 12, 13]. The second one, which is more challenging, concerns cohesive models and shells, for which ideas developped in [9, 12, 13], among others, are a good starting point.



Figure 15. Normalized  $K_1$  versus  $\mathcal{R}_J$  - J-integral - First test case.



Figure 16. Normalized  $K_1$  versus  $\mathcal{R}_J$  - J-integral - Second test case.

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